

XGM CONSULTING ENGINEERS

STRESS CALCULATION OF GATTI ICE CREAM 750 HEAT
EXCHANGER

ACCORDING TO PD5500 2018

DOCUMENT No. DC_2019-30-01

DRAWING No. 2019-30-01

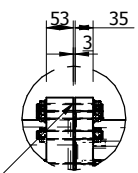
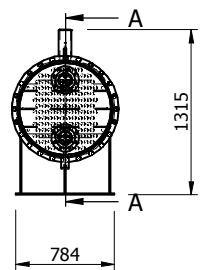
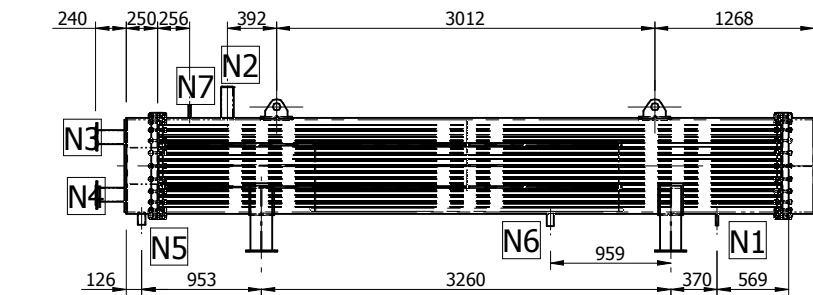
REPORT PREPARED FOR: CLAWS ENGINEERING

COMPILED BY: LE GREEN (PrTechEng)

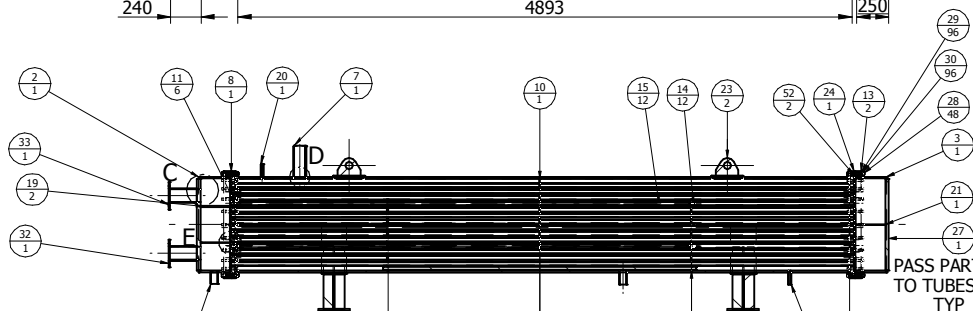
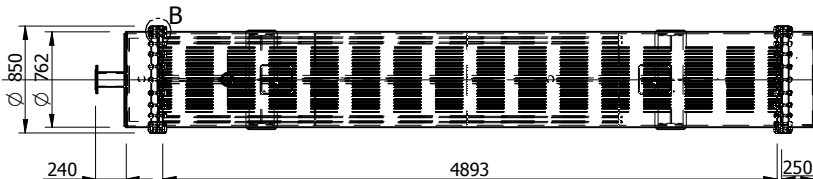


TÜV Rheinland
TUV JOB NO: 52070938
DATE: 15/01/20 SIGN: 
STATUS: A ☒ B ☐ C ☐ I/O ☐
ACTIVITY: DESIGN ☒ WELDING ☐ DESIGN REF: XG-ZN-04-12-19-1
QAWELDING REF: _____
A - PROCEED B - PROCEED, CHANGE AS NOTED & RESUBMIT
C - DO NOT PROCEED, CHANGE AS NOTED & RESUBMIT
I/O - NOT REVIEWED, FOR INFORMATION ONLY
Approval of this document does not imply approval of related documents.
Authorisation to proceed does not relieve the contractor/vendor of his
responsibility or liability under the contract/purchase order

DATE: 18 DECEMBER 2019



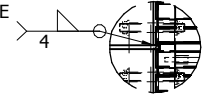
GASKET THICKNESS 3,17mm
DETAIL B



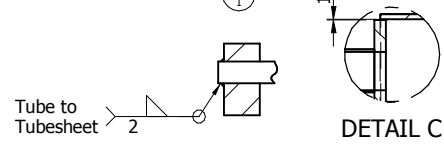
INSPECTION AUTHORITY	ZANGULA	
DESIGN CODE	PD5500 2018 CATEGORY III	
HAZARD CATEGORY (SANS 347)	III	
RADIOGRAPHY	NO	
DESIGN DATA		
	SHELL SIDE	TUBE SIDE
OPERATING PRESSURE	kPa 1200	900
DESIGN PRESSURE	kPa 1300	1000
HYDRAULIC TEST PRESSURE	kPa 1950	1500
TEST PROCEDURE	HYDRAULIC	HYDRAULIC
DESIGN FLOW	m ³ /s	
DESIGN TEMPERATURE MAX	(deg C) 35	35
DESIGN TEMPERATURE MIN	(deg C) -40	-40
CORROSION ALLOWANCE	(mm) 3.25	0
CUBIC CAPACITY	m ³ 9.5	
FLUID	AMMONIA	BRINE
	NO	NO

SIZE	RATING	WELD	MATERIAL	QTY
N1 32NB	3000 lb	WELD	A106B	AMMONIA INLET 1
N2 100NB	3000 lb	WELD	A106B	AMMONIA OUTLET 1
N3 100NB	SCH120	WELD	A106B	BRINE INLET 1
N4 100NB	SCH120	WELD	A106B	BRINE OUTLET 1
N5 50NB	3000 lb	WELD	A106B	TS DRAIN-PLUG 1
N6 50NB	3000 lb	WELD	A106B	SS DRAIN-PLUG 1
N7 15NB	6000 lb	WELD	A106B	PRESSURE GAUGE 1

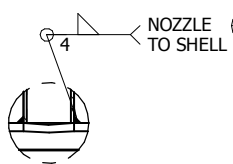
VESSEL DESIGNED ACCORDING TO PD5500 2018 CATIII
CONFORMITY ASSESMENT MODULE: B+C1
LESLIE GREEN (PrTechEng)
201470145



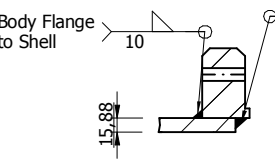
DETAIL E



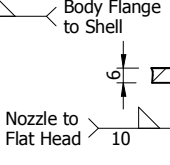
DETAIL C



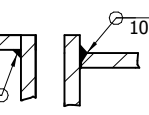
DETAIL D



Body Flange to Shell



Nozzle to Flat Head

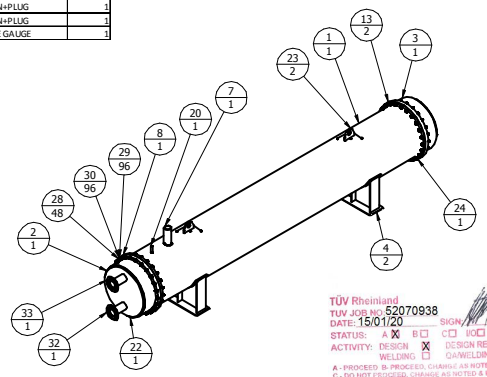


Tubesheet to Shell

REV	BY	CHKD	REVISION DESCRIPTION
1	R.H	L.G	CHANNELS BROUGHT TO 250mm

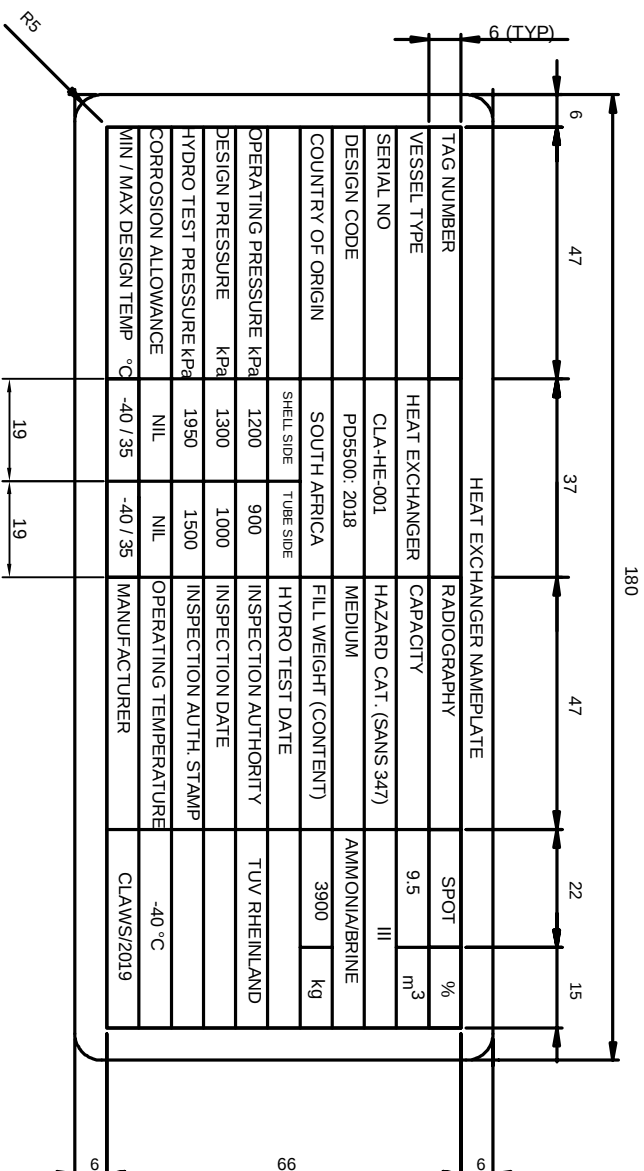


CHECKED	NAME	DATE	TITLE
L.G	L.G	18-12-19	SHELL AND HEAT EXCHANGER
DRAWN	NAME	DATE	TITLE
R.H	R.H	18-12-19	FOR APPROVAL
ENG APPR	NAME	DATE	TITLE
L.G	L.G		



TÜV Rheinland
TUV JOB No. 52070938
DATE: 15/01/20
STATUS: A X B C D E
ACTIVITY: DESIGN X
WELDING REF: XG-ZN-04-12-19-1
A - PROCESSED, B - PROCESSED, CHANGING AS REQUIRED & REDESIGN
C - NOT RECOMMENDED, CHANGING AS REQUIRED & REDESIGN
D - NOT RECOMMENDED, FOR INFORMATION ONLY
E - NOT RECOMMENDED, FOR INFORMATION ONLY
Approval of this document does not release the contractor of his responsibility for the design and construction of the vessel.

Item Number	Title	Quantity
1	Pipe_750NB_Sch30_4890mm_A 106B/ API 5LB	1
2	Pipe_750NB_Sch30_250mm_A 106B/ API 5LB	1
3	Pipe_750NB_Sch30_250mm_A 106B/ API 5LB	1
4	SADDLE_P355GH	2
7	N2 AMMONIA OUTLET NOZZLE_4"(100NB)_3000lb_A106B	1
8	TUBESHEET 35mm THK_LEFT SIDE_A-516 70	1
9*	BAFFLE BOTTOM PLATE_A-516 70	2
10*	BAFFLE TOP PLATE_A-516 70	1
11*	M10 TIE ROD_3660mm_BMS	6
12*	TUBE_OD 33.4_BWG 11_wt 3.38_4993mm_A-516 70	136
13	750mm BODY FLANGE_A516 Gr 70	2
14*	M10 HEX NUT	12
15*	SPACER_10NB OD17.2 WT 2.87_1204mm_A 106	12
16*	SPACER_10NB OD17.2 WT 2.87_1200mm_A 106	6
17*	BAFFLE STIFFENERS_SABS 62	1
18*	WELDED FLAT HEAD_PLATE 30mm THK OD 729mm_P355 NH	1
19*	PASS PARTITION PLATE LEFT_662x217x6mm_A-516 70	2
20	N7 PRESSURE GAUGE HALF COUPLING 1/2" 6000lb_A106B	1
21*	PASS PARTITION PLATE RIGHT_727x257x6mm_A-516 70	1
22	N5 TUBE SIDE DRAIN + PLUG_2"(50NB)_3000lb_A106B	1
23	LIFTING LUG_P355GH	2
24	TUBESHEET 35mm THK_RIGHT SIDE OD 850mm_A-516 70	1
26*	N1 AMMONIA INLET COUPLING_1 1/4" (32NB)_3000lb_A106B	1
27*	WELDED FLAT HEAD_PLATE 30mm THK OD 729mm_P355 NH	1
28	STUD BOLT M16 x 135mm_SA 193-B7	48
29	Washer M16	96
30	M16 NUT_SA 193-B7	96
31*	GASKET_OD790xID762_3.17mm_SF 5000 COMPRESSED FIBRE SHEET GASKET CARBON ARAMID NBR	2
32	N4 BRINE OUTLET_4"(100NB)-SCH120_A106B	1
33	N3 BRINE INLET_4"(100NB)-SCH120_A106B	1
34	N6 SHELL SIDE DRAIN + PLUG 2"(50NB)_3000lb_A106B	1



VESSEL DESIGNED ACCORDING
TO PD5500: 2018
LESLIE GREEN (PrTechEng)
201470150

CONFORMITY ASSESSMENT
MODULE A1

[Signature]

- NOTES:
1. HEIGHT OF ALL LETTERS TO BE 3 MM
 2. FOR ALL LETTERS USE "ARIAL NARROW" FONT
 3. FINAL ENGRAVING DETAIL FOR THE SPECIFIC VESSEL
WILL BE SUPPLIED WITH REQUEST OF QUOTATION
 4. TO BE STAMPED ON COMPLETION OF HYDRO TEST

TUV Rheinland
TUV JOB NO. 52070938
DATE: 15/01/20
STATUS: ☒ A ☐ B ☐ C ☐ D ☐ E ☐ F ☐ G ☐ H ☐ I ☐ J ☐ K ☐ L ☐ M ☐ N ☐ O ☐ P ☐ Q ☐ R ☐ S ☐ T ☐ U ☐ V ☐ W ☐ X ☐ Y ☐ Z
ACTIVITY: DESIGN ☒ QAWELDING REF: XG-ZN-04-12-19-1
WELDING ☐ QAWELDING REF: XG-ZN-04-12-19-1
A - PROCEED B - PROCEED, CHANGE AS NOTED & RESUBMIT
C - DO NOT PROCEED, CHANGE AS NOTED & RESUBMIT
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Authorisation to produce this document does not imply the responsibility of related documents
Authorisation to produce this document does not imply the responsibility of related documents

TITLE				DOCUMENT TYPE	REVISION
Name Plate_ HEAT EXCHANGER				Detail part drawing	1.00
	NAME	SIGNATURE	DATE	DWG NO.	
DRAWN	R.H	L.G	2019-06-26	300-2019-GIC-S&T_001	
CHK'D	L.G	L.G			
APPR'D	L.G	L.G			
	GATTI ICE CREAM			MATERIAL	
	SHELL & TUBE HE				
	SCALE:1				
	Page size: A3		SHEET 1 OF 1		

PV Elite Vessel Analysis Program: Input Data

Exchanger Design Pressures and Temperatures

Shell Side Design Pressure	1.3000	N/mm ²
Channel Side Design Pressure	1.0000	N/mm ²
Shell Side Design Temperature	-40	°C
Channel Side Design Temperature	-40	°C

Hydrotest Position	Horizontal
Projection of Nozzle from Vessel Top	0.0000 mm
Projection of Nozzle from Vessel Bottom	0.0000 mm
Miscellaneous Weight Percent	0.0
Use Higher Longitudinal Stresses (Flag)	Y
Select t for Internal Pressure (Flag)	N
Select t for External Pressure (Flag)	N
Select t for Axial Stress (Flag)	N
Consider Vortex Shedding	Y
Perform a Corroded Hydrotest	Y
Is this a Heat Exchanger	No
User Defined Hydro. Press. (Used if > 0)	0.0000 N/mm ²
User defined MAWP	0.0000 N/mm ²
User defined MAPnc	0.0000 N/mm ²

Load Case 1	NP+EW+WI+FW+BW
Load Case 2	NP+EW+EE+FS+BS
Load Case 3	NP+OW+WI+FW+BW
Load Case 4	NP+OW+EQ+FS+BS
Load Case 5	NP+HW+HI
Load Case 6	NP+HW+HE
Load Case 7	IP+OW+WI+FW+BW
Load Case 8	IP+OW+EQ+FS+BS
Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS

FileName : Gatti 750 heat exchanger

Input Echo :

Step: 1 10:06a Dec 18, 2019

Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+EW
Load Case 17	NP+VO+OW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

Wind Design Code	ASCE-7 93
Basic Wind Speed	[V] 31.292 m/sec
Surface Roughness Category	C: Open Terrain
Importance Factor	1.0
Type of Surface	Moderately Smooth
Base Elevation	0.0000 m
Percent Wind for Hydrotest	33.0
Using User defined Wind Press. Vs Elev.	N
Damping Factor (Beta) for Wind (Ope)	0.0100
Damping Factor (Beta) for Wind (Empty)	0.0000
Damping Factor (Beta) for Wind (Filled)	0.0000

Seismic Design Code	UBC 94
UBC Seismic Zone (1=1,2=2a,3=2b,4=3,5=4)	0.000
UBC Importance Factor	1.000
UBC Soil Type	S1
UBC Horizontal Force Factor	3.000
UBC Percent Seismic for Hydrotest	0.000

Design Nozzle for Des. Press. + St. Head	Y
Consider MAP New and Cold in Noz. Design	N
Consider External Loads for Nozzle Des.	Y
Use ASME VIII-1 Appendix 1-9	N

Complete Listing of Vessel Elements and Details:

Element From Node	10	
Element To Node	20	
Element Type	Flat	
Description	Left Head	
Distance "FROM" to "TO"	0.03000	m
Element Outside Diameter	734.00	mm
Element Thickness	30.000	mm
Internal Corrosion Allowance	3.1750	mm
Nominal Thickness	30.000	mm
External Corrosion Allowance	0.0000	mm
Design Internal Pressure	1.0000	N/mm ²
Design Temperature Internal Pressure	-40	°C
Design External Pressure	0.1030	N/mm ²
Design Temperature External Pressure	0	°C
Effective Diameter Multiplier	1.2	
Material Name	P355NH	
Allowable Stress, Ambient	191485.	kPa
Allowable Stress, Operating	191485.	kPa
Material Density	7810.0	kg/m ³
Flat Head Attachment Factor	0.41	
Small diameter if Non-Circular	0.0000	mm

Element From Node	10	
Detail Type	Nozzle	
Detail ID	Noz N1 Fr10	
Dist. from "FROM" Node / Offset dist	200.00	mm
Nozzle Diameter	100.0	mm
Nozzle Schedule	120	
Nozzle Class	150	
Layout Angle	270.0	
Blind Flange (Y/N)	N	
Weight of Nozzle (Used if > 0)	0.0000	N
Grade of Attached Flange	GR 1.1	
Nozzle Matl	A-106 B	

FileName : Gatti 750 heat exchanger

Input Echo :

Step: 1 10:06a Dec 18, 2019

Element From Node	10
Detail Type	Nozzle
Detail ID	Noz N2 Fr10
Dist. from "FROM" Node / Offset dist	200.00 mm
Nozzle Diameter	100.0 mm
Nozzle Schedule	120
Nozzle Class	150
Layout Angle	90.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	461.84 N
Grade of Attached Flange	GR 1.1
Nozzle Matl	A-106 B

Element From Node	20
Element To Node	30
Element Type	Cylinder
Description	Left Channel Shell
Distance "FROM" to "TO"	0.2500 m
Element Outside Diameter	762.00 mm
Element Thickness	13.891 mm
Internal Corrosion Allowance	3.1750 mm
Nominal Thickness	13.896 mm
External Corrosion Allowance	0.0000 mm
Design Internal Pressure	1.0000 N/mm ²
Design Temperature Internal Pressure	-40 °C
Design External Pressure	0.1030 N/mm ²
Design Temperature External Pressure	35 °C
Effective Diameter Multiplier	1.2
Material Name	A-106 B
Allowable Stress, Ambient	159996. kPa
Allowable Stress, Operating	159996. kPa
Material Density	7810.0 kg/m ³

FileName : Gatti 750 heat exchanger

Input Echo :

Step: 1 10:06a Dec 18,2019

Element From Node	20
Detail Type	Nozzle
Detail ID	Noz N1 Fr20
Dist. from "FROM" Node / Offset dist	0.1400 m
Nozzle Diameter	76.199997 mm
Nozzle Schedule	None
Nozzle Class	None
Layout Angle	270.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.0000 N
Grade of Attached Flange	None
Nozzle Matl	A-106 B

Element From Node	30
Element To Node	40
Element Type	Flange
Description	Flange 30
Distance "FROM" to "TO"	0.05500 m
Flange Inside Diameter	0.0000 mm
Element Thickness	55.000 mm
Internal Corrosion Allowance	3.1750 mm
Nominal Thickness	0.0000 mm
External Corrosion Allowance	0.0000 mm
Design Internal Pressure	1.0000 N/mm ²
Design Temperature Internal Pressure	-40 °C
Design External Pressure	0.1030 N/mm ²
Design Temperature External Pressure	-40 °C
Effective Diameter Multiplier	1.2
Material Name	SA-516 70
Allowable Stress, Ambient	153188. kPa
Allowable Stress, Operating	153188. kPa
Material Density	7840.0 kg/m ³
Perform Flange Stress Calculation (Y/N)	Y
Weight of ANSI B16.5/B16.47 Flange	0.0000 N

Step: 1 10:06a Dec 18,2019

Class of ANSI B16.5/B16.47 Flange

Grade of ANSI B16.5/B16.47 Flange

Element From Node	40	
Element To Node	50	
Element Type	Cylinder	
Description	Main Shell	
Distance "FROM" to "TO"	4.8400	m
Element Outside Diameter	762.00	mm
Element Thickness	13.891	mm
Internal Corrosion Allowance	3.1750	mm
Nominal Thickness	13.896	mm
External Corrosion Allowance	0.0000	mm
Design Internal Pressure	1.3000	N/mm ²
Design Temperature Internal Pressure	-40	°C
Design External Pressure	0.1030	N/mm ²
Design Temperature External Pressure	-40	°C
Effective Diameter Multiplier	1.2	
Material Name	A-106 B	
Allowable Stress, Ambient	159996.	kPa
Allowable Stress, Operating	159996.	kPa
Material Density	7810.0	kg/m ³
Element From Node	40	
Detail Type	Saddle	
Detail ID	New Sdl	
Dist. from "FROM" Node / Offset dist	4.2000	m
Width of Saddle	204.00	mm
Height of Saddle at Bottom	572.00	mm
Saddle Contact Angle	120.0	
Height of Composite Ring Stiffener	0.0000	mm
Width of Wear Plate	306.00	mm
Thickness of Wear Plate	10.000	mm
Contact Angle, Wear Plate (degrees)	132.0	

Element From Node	40
Detail Type	Saddle
Detail ID	Sdl 2 Fr40
Dist. from "FROM" Node / Offset dist	0.7000 m
Width of Saddle	204.00 mm
Height of Saddle at Bottom	572.00 mm
Saddle Contact Angle	120.0
Height of Composite Ring Stiffener	0.0000 mm
Width of Wear Plate	306.00 mm
Thickness of Wear Plate	10.000 mm
Contact Angle, Wear Plate (degrees)	132.0

Element From Node	40
Detail Type	Nozzle
Detail ID	Noz N1 Fr40
Dist. from "FROM" Node / Offset dist	0.2000 m
Nozzle Diameter	139.7 mm
Nozzle Schedule	None
Nozzle Class	None
Layout Angle	90.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	279.80 N
Grade of Attached Flange	None
Nozzle Matl	A-106 B

Element From Node	40
Detail Type	Nozzle
Detail ID	Noz N2 Fr40
Dist. from "FROM" Node / Offset dist	0.4500 m
Nozzle Diameter	38.099998 mm
Nozzle Schedule	None
Nozzle Class	None
Layout Angle	90.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.0000 N

Step: 1 10:06a Dec 18,2019

Grade of Attached Flange	None
Nozzle Matl	A-106 B
Element From Node	40
Detail Type	Nozzle
Detail ID	Noz N3 Fr40
Dist. from "FROM" Node / Offset dist	2.5000 m
Nozzle Diameter	76.199997 mm
Nozzle Schedule	None
Nozzle Class	None
Layout Angle	270.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.0000 N
Grade of Attached Flange	None
Nozzle Matl	A-106 B

Element From Node	40
Detail Type	Nozzle
Detail ID	Noz N4 Fr40
Dist. from "FROM" Node / Offset dist	4.5500 m
Nozzle Diameter	57.150002 mm
Nozzle Schedule	None
Nozzle Class	None
Layout Angle	270.0
Blind Flange (Y/N)	N
Weight of Nozzle (Used if > 0)	0.0000 N
Grade of Attached Flange	None
Nozzle Matl	A-106 B

Element From Node	50
Element To Node	60
Element Type	Flange
Description	Flange 50
Distance "FROM" to "TO"	0.05500 m

FileName : Gatti 750 heat exchanger

Input Echo :

Step: 1 10:06a Dec 18,2019

Flange Inside Diameter	748.13	mm
Element Thickness	55.000	mm
Internal Corrosion Allowance	3.1750	mm
Nominal Thickness	0.0000	mm
External Corrosion Allowance	0.0000	mm
Design Internal Pressure	1.0000	N/mm ²
Design Temperature Internal Pressure	40	°C
Design External Pressure	0.1030	N/mm ²
Design Temperature External Pressure	35	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-516	70
Allowable Stress, Ambient	153188.	kPa
Allowable Stress, Operating	153188.	kPa
Material Density	7840.0	kg/m ³
Perform Flange Stress Calculation (Y/N)	Y	
Weight of ANSI B16.5/B16.47 Flange	0.0000	N
Class of ANSI B16.5/B16.47 Flange		
Grade of ANSI B16.5/B16.47 Flange		

Element From Node	60	
Element To Node	70	
Element Type	Cylinder	
Description	Right Channel	
Distance "FROM" to "TO"	0.2500	m
Element Outside Diameter	762.00	mm
Element Thickness	13.891	mm
Internal Corrosion Allowance	3.1750	mm
Nominal Thickness	13.896	mm
External Corrosion Allowance	0.0000	mm
Design Internal Pressure	1.0000	N/mm ²
Design Temperature Internal Pressure	-40	°C
Design External Pressure	0.1030	N/mm ²
Design Temperature External Pressure	-40	°C
Effective Diameter Multiplier	1.2	

Step: 1 10:06a Dec 18, 2019

Material Name	A-106 B
Allowable Stress, Ambient	159996. kPa
Allowable Stress, Operating	159996. kPa
Material Density	7810.0 kg/m³

Element From Node	70
Element To Node	80
Element Type	Flat
Description	Right head
Distance "FROM" to "TO"	0.03000 m
Element Outside Diameter	733.00 mm
Element Thickness	30.000 mm
Internal Corrosion Allowance	3.1750 mm
Nominal Thickness	30.000 mm
External Corrosion Allowance	0.0000 mm
Design Internal Pressure	1.0000 N/mm²
Design Temperature Internal Pressure	-40 °C
Design External Pressure	0.1030 N/mm²
Design Temperature External Pressure	-40 °C
Effective Diameter Multiplier	1.2
Material Name	P355NH
Allowable Stress, Ambient	191485. kPa
Allowable Stress, Operating	191485. kPa
Material Density	7810.0 kg/m³
Flat Head Attachment Factor	0.41
Small diameter if Non-Circular	0.0000 mm

XY Coordinate Calculations

From	To	X (Horiz.)	Y (Vert.)	DX (Horiz.)	DY (Vert.)
		m	m	m	m

Left Head		0.030000	...	0.030000	...
Left Chann		0.28000	...	0.25000	...
Flange 30		0.28000	...	-0.055000	...
Main Shell		5.15600	...	4.84000	...
Flange 50		5.24700	...	0.055000	...
Right Chan		5.44200	...	0.25000	...
Right head		5.47200	...	0.030000	...

Flange Input Data Values

Description: New Flange :

Flange 30

Description of Flange Geometry (Type)		Integral Ring	
Design Pressure	P	1.00	N/mm ²
Design Temperature		-40	°C
Internal Corrosion Allowance	ci	3.1750	mm
External Corrosion Allowance	ce	0.0000	mm
Use Corrosion Allowance in Thickness Calcs.		No	
Attached Shell Inside Diameter	B	762.0000	mm
Integral Ring Inside Diameter		789.7920	mm
Flange Outside Diameter	A	850.000	mm
Flange Thickness	t	55.0000	mm
Flange Material		SA-516 70	
Flange Allowable Stress At Temperature	Sfo	153188.00	kPa
Flange Allowable Stress At Ambient	Sfa	153188.00	kPa
Bolt Material		42CrMo5-6	
Bolt Allowable Stress At Temperature	Sb	226994.81	kPa
Bolt Allowable Stress At Ambient	Sa	226994.81	kPa
Length of Weld Leg at Back of Ring	tw	0.0000	mm
Number of Splits in Ring Flange	n	0	
Diameter of Bolt Circle	C	810.000	mm
Nominal Bolt Diameter	db	16.0000	mm
Diameter of Bolt Hole	dh	19.0000	mm
Type of Threads		TEMA Metric	
Number of Bolts		24	
Flange Face Outside Diameter	Fod	790.000	mm
Flange Face Inside Diameter	Fid	762.000	mm

Flng: 3 10:06a Dec 18,2019

Flange Facing Sketch	1, Code Sketch 1a
Gasket Outside Diameter	Go 790.000 mm
Gasket Inside Diameter	Gi 762.000 mm
Gasket Factor	m 3.7000
Gasket Design Seating Stress	y 22993.50 kPa
Column for Gasket Seating	2, Code Column II
Gasket Thickness	tg 3.1750 mm

PD 5500:2018

Corroded Flange ID,	$B_{cor} = B + 2 * F_{cor}$	768.350 mm
Corroded Large Hub,	$g1_{cor} = g1 - ci$	0.000 mm
Corroded Small Hub,	$g0_{cor} = g0 - ci$	0.000 mm
Code R Dimension,	$R = ((C - B_{cor}) / 2) - g1_{cor}$	20.825 mm
Gasket Contact Width,	$N = (Go - Gi) / 2$	10.825 mm
Basic Gasket Width,	$bo = N / 2$	5.413 mm
Effective Gasket Width,	$b = bo$	5.413 mm
Gasket Reaction Diameter,	$G = (Go + Gi) / 2$	776.000 mm

Basic Flange and Bolt Loads:

Hydrostatic End Load due to Pressure [H]:

$$\begin{aligned}
 &= 0.785 * G^2 * P_{eq} \\
 &= 0.785 * 776.0000^2 * 1.000 \\
 &= 472907.719 \text{ N}
 \end{aligned}$$

Contact Load on Gasket Surfaces [Hp]:

$$\begin{aligned}
 &= 2 * b * P_i * G * m * P \\
 &= 2 * 5.4125 * 3.1416 * 776.0000 * 3.7000 * 1.00 \\
 &= 97634.875 \text{ N}
 \end{aligned}$$

Hydrostatic End Load at Flange ID [Hd]:

$$\begin{aligned}
 &= P_i * B_{cor}^2 * P / 4 \\
 &= 3.1416 * 768.3500^2 * 1.0000 / 4 \\
 &= 463629.562 \text{ N}
 \end{aligned}$$

Pressure Force on Flange Face [Ht]:

= H - Hd
= 472907 - 463629
= 9278.146 N

Operating Bolt Load [Wm1]:

= max(H + Hg + Hr, 0)
= max(472907 + 97634 + 0 , 0)
= 570542.562 N

Gasket Seating Bolt Load [Wm2]:

= Pi * G * b' * y + yPart * bPart * lp
= 3.141*776.00*5.4125*22993.500+0.00*0.0000*0.00
= 303391.656 N

Required Bolt Area [Am]:

= Maximum of Wm1/Sb, Wm2/Sa
= Maximum of 570542/226994 , 303391/226994
= 0.003 m²

Bolt Spacing Correction Factor per TEMA, PD:5500 or EN-13445 [cF]:

= max(sqrt(Deltab /(2 * dB + 6 * e / (m + 0.5))); 1.0)
= max(sqrt(105.726/(2 * 16.000 + 6 * 55.000/(3.700 + 0.5))); 1.0)
= max(0.9778 ; 1.0)
= 1.0000

where the actual circumferential spacing is [Deltab]:

= C * sin(Pi / n))
= 810.000 * sin(3.142/24)
= 105.7262 mm

Bolting Information for TEMA Metric Thread Series (Non Mandatory):

Distance Across Corners for Nuts		31.180	mm
Circular Wrench End Diameter	a	0.000	mm

	Minimum	Actual	Maximum

Bolt Area, m²	0.003	0.003	
Radial distance bet. hub and bolts	20.640	24.000	

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : New Flange

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Radial distance bet. bolts and the edge 20.640 20.000

Circumferential spacing between bolts 44.450 105.726 110.571

Min. Gasket Contact Width (Brownell Young) [Not an ASME Calc] [Nmin]:

$$\begin{aligned}
 &= A_b * S_a / (y * \pi * (G_o + G_i)) \\
 &= 0.003 * 226994.81 / (22993.50 * 3.14 * (790.000 + 762.00)) \\
 &= 6.722 \text{ mm}
 \end{aligned}$$

Flange Design Bolt Load, Gasket Seating [W]:

$$\begin{aligned}
 &= S_a * A_b \\
 &= 226994.81 * 0.0033 \\
 &= 753551.69 \text{ N}
 \end{aligned}$$

Gasket Load for the Operating Condition [HG]:

$$\begin{aligned}
 &= W_{m1} - H \\
 &= 570542 - 472907 \\
 &= 97634.88 \text{ N}
 \end{aligned}$$

Moment Arm Calculations:

Distance to Gasket Load Reaction [hg]:

$$\begin{aligned}
 &= (C - G) / 2 \\
 &= (810.0000 - 776.0000) / 2 \\
 &= 17.0000 \text{ mm}
 \end{aligned}$$

Distance to Face Pressure Reaction [ht]:

$$\begin{aligned}
 &= (R + g_1 + h_g) / 2 \\
 &= (20.8250 + 0.0000 + 17.0000) / 2 \\
 &= 18.9125 \text{ mm}
 \end{aligned}$$

Distance to End Pressure Reaction [hd]:

$$\begin{aligned}
 &= R + (g_1 / 2) \\
 &= 20.8250 + (0.0000 / 2.0) \\
 &= 20.8250 \text{ mm}
 \end{aligned}$$

Summary of Moments for Internal Pressure:

Loading		Force	Distance	Bolt Corr	Moment
End Pressure,	Md	463630.	20.8250	1.0000	9659. N-m
Face Pressure,	Mt	9278.	18.9125	1.0000	176. N-m

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : New Flange

Flng: 3 10:06a Dec 18, 2019

Gasket Load, Mg 97635. 17.0000 1.0000 1660. N-m

Gasket Seating, Matm 753552. 17.0000 1.0000 12816. N-m

Total Moment for Operation, Mop 11495. N-m

Total Moment for Gasket seating, Matm 12816. N-m

Effective Hub Length, ho = 0.000 mm

Hub Ratio, h/h0 = Defined as 0.0 0.000

Thickness Ratio, g1/g0 = Defined as 0.0 0.000

Factors from Figure 2-7.1 K = 1.106

T = 1.875 U = 21.090

Y = 19.192 Z = 9.936

Tangential Flange Stress, Operating [Sto]:

$$= (Y * Mo) / (t^2 * Bcor)$$

$$= (19.1918 * 11495) / (55.0000^2 * 768.3500)$$

$$= 94880.33 \text{ kPa}$$

Tangential Flange Stress, Seating [STa]:

$$= (Y * Matm) / (t^2 * Bcor)$$

$$= (19.1918 * 12815) / (55.0000^2 * 768.3500)$$

$$= 105780.27 \text{ kPa}$$

Bolt Stress, Operating [BSo]:

$$= (Wm1 / Ab)$$

$$= (570542 / 0.0033)$$

$$= 171866.41 \text{ kPa}$$

Bolt Stress, Seating [BSa]:

$$= (Wm2 / Ab)$$

$$= (303391 / 0.0033)$$

$$= 91391.66 \text{ kPa}$$

Flange Stress Analysis Results:

	Operating		Gasket Seating	
	Actual	Allowed	Actual	Allowed

Tangential Flange	94880.	153188.	105780.	153188. kPa

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : New Flange

Flng: 3 10:06a Dec 18, 2019

Bolting	171866.	226995.	91392.	226995.	kPa
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Minimum Required Flange Thickness

46.533 mm

Estimated M.A.W.P. (Operating)

1.321 N/mm²

Estimated Finished Weight of Flange at given Thk.

48.0 kgm

Estimated Unfinished Weight of Forging at given Thk

48.0 kgm

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Flange Input Data Values

Description: FLANGE :

Flange 50

Description of Flange Geometry (Type)		Integral Ring	
Design Pressure	P	1.00	N/mm ²
Design Temperature		40	°C
Internal Corrosion Allowance	ci	3.1750	mm
External Corrosion Allowance	ce	0.0000	mm
Use Corrosion Allowance in Thickness Calcs.		No	
Attached Shell Inside Diameter	B	762.0000	mm
Integral Ring Inside Diameter		789.7920	mm
Flange Outside Diameter	A	850.000	mm
Flange Thickness	t	55.0000	mm
Flange Material		SA-516 70	
Flange Allowable Stress At Temperature	Sfo	153187.98	kPa
Flange Allowable Stress At Ambient	Sfa	153187.98	kPa
Bolt Material		42CrMo5-6	
Bolt Allowable Stress At Temperature	Sb	226994.81	kPa
Bolt Allowable Stress At Ambient	Sa	226994.81	kPa
Length of Weld Leg at Back of Ring	tw	0.0000	mm
Number of Splits in Ring Flange	n	0	
Diameter of Bolt Circle	C	810.000	mm
Nominal Bolt Diameter	db	16.0000	mm
Diameter of Bolt Hole	dh	19.0000	mm
Type of Threads		TEMA Metric	
Number of Bolts		24	
Flange Face Outside Diameter	Fod	790.000	mm
Flange Face Inside Diameter	Fid	762.000	mm

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Flange Facing Sketch	1, Code Sketch 1a
Gasket Outside Diameter	Go 790.000 mm
Gasket Inside Diameter	Gi 762.000 mm
Gasket Factor	m 3.7000
Gasket Design Seating Stress	y 22993.50 kPa
Column for Gasket Seating	2, Code Column II
Gasket Thickness	tg 3.1750 mm

PD 5500:2018

Corroded Flange ID,	$B_{cor} = B + 2 * F_{cor}$	768.350 mm
Corroded Large Hub,	$g1_{cor} = g1 - ci$	0.000 mm
Corroded Small Hub,	$g0_{cor} = g0 - ci$	0.000 mm
Code R Dimension,	$R = ((C - B_{cor}) / 2) - g1_{cor}$	20.825 mm
Gasket Contact Width,	$N = (Go - Gi) / 2$	10.825 mm
Basic Gasket Width,	$bo = N / 2$	5.413 mm
Effective Gasket Width,	$b = bo$	5.413 mm
Gasket Reaction Diameter,	$G = (Go + Gi) / 2$	776.000 mm

Basic Flange and Bolt Loads:

Hydrostatic End Load due to Pressure [H]:

$$\begin{aligned}
 &= 0.785 * G^2 * P_{eq} \\
 &= 0.785 * 776.0000^2 * 1.000 \\
 &= 472907.719 \text{ N}
 \end{aligned}$$

Contact Load on Gasket Surfaces [Hp]:

$$\begin{aligned}
 &= 2 * b * P_i * G * m * P \\
 &= 2 * 5.4125 * 3.1416 * 776.0000 * 3.7000 * 1.00 \\
 &= 97634.875 \text{ N}
 \end{aligned}$$

Hydrostatic End Load at Flange ID [Hd]:

$$\begin{aligned}
 &= P_i * B_{cor}^2 * P / 4 \\
 &= 3.1416 * 768.3500^2 * 1.0000 / 4 \\
 &= 463629.562 \text{ N}
 \end{aligned}$$

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : FLANGE

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Pressure Force on Flange Face [Ht]:

$$= H - H_d$$

$$= 472907 - 463629$$

$$= 9278.146 \text{ N}$$

Operating Bolt Load [Wm1]:

$$= \max(H + H_g + H_r, 0)$$

$$= \max(472907 + 97634 + 0, 0)$$

$$= 570542.562 \text{ N}$$

Gasket Seating Bolt Load [Wm2]:

$$= P_i * G * b' * y + y_{Part} * b_{Part} * l_p$$

$$= 3.141 * 776.00 * 5.4125 * 22993.500 + 0.00 * 0.0000 * 0.00$$

$$= 303391.656 \text{ N}$$

Required Bolt Area [Am]:

$$= \text{Maximum of } Wm1/S_b, Wm2/S_a$$

$$= \text{Maximum of } 570542/226994, 303391/226994$$

$$= 0.003 \text{ m}^2$$

Bolt Spacing Correction Factor per TEMA, PD:5500 or EN-13445 [cF]:

$$= \max(\sqrt{\text{Deltab} / (2 * d_B + 6 * e / (m + 0.5))}, 1.0)$$

$$= \max(\sqrt{105.726 / (2 * 16.000 + 6 * 55.000 / (3.700 + 0.5))}, 1.0)$$

$$= \max(0.9778, 1.0)$$

$$= 1.0000$$

where the actual circumferential spacing is [Deltab]:

$$= C * \sin(\pi / n)$$

$$= 810.000 * \sin(3.142/24)$$

$$= 105.7262 \text{ mm}$$

Bolting Information for TEMA Metric Thread Series (Non Mandatory):

Distance Across Corners for Nuts 31.180 mm

Circular Wrench End Diameter a 0.000 mm

	Minimum	Actual	Maximum
Bolt Area, m ²	0.003	0.003	
Radial distance bet. hub and bolts	20.640	24.000	

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : FLANGE

Flng: 4 10:06a Dec 18, 2019

Radial distance bet. bolts and the edge 20.640 20.000

Circumferential spacing between bolts 44.450 105.726 110.571

Min. Gasket Contact Width (Brownell Young) [Not an ASME Calc] [Nmin]:

$$\begin{aligned}
 &= A_b * S_a / (y * \pi * (G_o + G_i)) \\
 &= 0.003 * 226994.81 / (22993.50 * 3.14 * (790.000 + 762.00)) \\
 &= 6.722 \text{ mm}
 \end{aligned}$$

Flange Design Bolt Load, Gasket Seating [W]:

$$\begin{aligned}
 &= S_a * A_b \\
 &= 226994.81 * 0.0033 \\
 &= 753551.69 \text{ N}
 \end{aligned}$$

Gasket Load for the Operating Condition [HG]:

$$\begin{aligned}
 &= W_{m1} - H \\
 &= 570542 - 472907 \\
 &= 97634.88 \text{ N}
 \end{aligned}$$

Moment Arm Calculations:

Distance to Gasket Load Reaction [hg]:

$$\begin{aligned}
 &= (C - G) / 2 \\
 &= (810.0000 - 776.0000) / 2 \\
 &= 17.0000 \text{ mm}
 \end{aligned}$$

Distance to Face Pressure Reaction [ht]:

$$\begin{aligned}
 &= (R + g_1 + h_g) / 2 \\
 &= (20.8250 + 0.0000 + 17.0000) / 2 \\
 &= 18.9125 \text{ mm}
 \end{aligned}$$

Distance to End Pressure Reaction [hd]:

$$\begin{aligned}
 &= R + (g_1 / 2) \\
 &= 20.8250 + (0.0000 / 2.0) \\
 &= 20.8250 \text{ mm}
 \end{aligned}$$

Summary of Moments for Internal Pressure:

Loading		Force	Distance	Bolt Corr	Moment
End Pressure,	Md	463630.	20.8250	1.0000	9659. N-m
Face Pressure,	Mt	9278.	18.9125	1.0000	176. N-m

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : FLANGE

Flng: 4 10:06a Dec 18, 2019

Gasket Load, Mg 97635. 17.0000 1.0000 1660. N-m

Gasket Seating, Matm 753552. 17.0000 1.0000 12816. N-m

Total Moment for Operation, Mop 11495. N-m

Total Moment for Gasket seating, Matm 12816. N-m

Effective Hub Length, ho = 0.000 mm

Hub Ratio, h/h0 = Defined as 0.0 0.000

Thickness Ratio, g1/g0 = Defined as 0.0 0.000

Factors from Figure 2-7.1 K = 1.106

T = 1.875 U = 21.090

Y = 19.192 Z = 9.936

Tangential Flange Stress, Operating [Sto]:

$$= (Y * Mo) / (t^2 * Bcor)$$

$$= (19.1918 * 11495) / (55.0000^2 * 768.3500)$$

$$= 94880.33 \text{ kPa}$$

Tangential Flange Stress, Seating [STa]:

$$= (Y * Matm) / (t^2 * Bcor)$$

$$= (19.1918 * 12815) / (55.0000^2 * 768.3500)$$

$$= 105780.27 \text{ kPa}$$

Bolt Stress, Operating [BSo]:

$$= (Wm1 / Ab)$$

$$= (570542 / 0.0033)$$

$$= 171866.41 \text{ kPa}$$

Bolt Stress, Seating [BSa]:

$$= (Wm2 / Ab)$$

$$= (303391 / 0.0033)$$

$$= 91391.66 \text{ kPa}$$

Flange Stress Analysis Results:

	Operating		Gasket Seating	
	Actual	Allowed	Actual	Allowed

Tangential Flange	94880.	153188.	105780.	153188. kPa

FileName : Gatti 750 heat exchanger

Flg Calc [Int P] : FLANGE

Flng: 4 10:06a Dec 18, 2019

Bolting	171866.	226995.	91392.	226995.	kPa
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Minimum Required Flange Thickness	46.533	mm
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Estimated M.A.W.P. (Operating)	1.321	N/mm ²
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Estimated Finished Weight of Flange at given Thk.	48.0	kgm
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Estimated Unfinished Weight of Forging at given Thk	48.0	kgm
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Element Thickness, Pressure, Diameter and Allowable Stress :

From	To	Int. Press + Liq. Hd N/mm ²	Nominal Thickness mm	Total Corr Allowance mm	Element Diameter mm	Allowable Stress(SE) kPa

Left Head		1.0000	30.000	3.1750	734.00	191485.
Left Chann		1.0000	13.896	3.1750	762.00	159996.
Flange 30		1.0000	...	3.1750	...	153188.
Main Shell		1.3000	13.896	3.1750	762.00	159996.
Flange 50		1.0000	...	3.1750	748.13	153188.
Right Chan		1.0000	13.896	3.1750	762.00	159996.
Right head		1.0000	30.000	3.1750	733.00	191485.

Element Required Thickness and MAWP :

From	To	Design Pressure N/mm ²	M.A.W.P. Corroded N/mm ²	M.A.P. New & Cold N/mm ²	Minimum Thickness mm	Required Thickness mm

Left Head		1.00000	1.87051	1.90302	30.0000	25.1102
Left Chann		1.00000	4.56434	5.94185	13.8906	5.54875
Flange 30		1.00000	1.32074	1.26840	55.0000	46.5328
Main Shell		1.30000	4.56434	5.94185	13.8906	6.25799
Flange 50		1.00000	1.32074	1.26840	55.0000	46.5328
Right Chan		1.00000	4.56434	5.94185	13.8906	5.54875
Right head		1.00000	1.87558	1.90821	30.0000	25.0805

Summary of Heat Exchanger Maximum Allowable Working Pressures :

Note: The MAWPs and MAPs for the Exchanger elements all appear to

be zero or some could not be calculated. Please fill these

values in on the main tubesheet tab.

Internal Pressure Calculation Results :

British Standard PD 5500:2018

Welded Flat Head From 10 To 20 P355NH at -40 °C

Left Head

Thickness Due to Internal Pressure (er):

$$\begin{aligned}
 &= C * D * \sqrt{p/f} \text{ per 3.5.5-1} \\
 &= 0.4100 * 740.35 * \sqrt{1.00/191484.98} \\
 &= 21.9352 + 3.1750 = 25.1102 \text{ mm}
 \end{aligned}$$

Max. All. Working Pressure at Given Thickness (MAWP):

$$\begin{aligned}
 &= f * (e_{cor} / (C * D_{ca}))^2 \\
 &= 191484.98 * (26.83 / (0.41 * 740.35))^2 \\
 &= 1.87 \text{ N/mm}^2
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold (MAPNC):

$$\begin{aligned}
 &= f_a * (e / (C * D))^2 \\
 &= 191484.98 * (30.00 / (0.41 * 734.00))^2 \\
 &= 1.90 \text{ N/mm}^2
 \end{aligned}$$

Actual Stress at Given Thickness:

$$= 102370.34 \text{ kPa}$$

Note : For flat heads with openings, C should be the Maximum of CY1 and
 $0.41 * Y2$ per 3.5.5-5. $d/D \leq 0.5$.

NOTE: If there are nozzles in the Flat Head,

check the Nozzle Summary for Final Flat Head Thickness
 requirement.

As the Minimum Design Temperature is: 50.0 °C , Impact is Not Required
 per Annex D

Cylindrical Shell From 20 To 30 A-106 B at -40 °C

Left Channel Shell

Thickness Due to Internal Pressure (er):

$$\begin{aligned}
 &= (P * Do) / (2 * f + p) \text{ per 3.5.1.2 Eqn. 3.5.1-2} \\
 &= (1.00*762.0000)/(2*159996.34+1.00) \\
 &= 2.3737 + 3.1750 = 5.5487 \text{ mm}
 \end{aligned}$$

Max. All. Working Pressure at Given Thickness (MAWP):

$$\begin{aligned}
 &= (2 * f * e_{cor}) / (Do_{ca} - e_{cor}) \\
 &= (2*159996.34*10.72)/(762.00-10.72) \\
 &= 4.56 \text{ N/mm}^2
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold (MAPNC):

$$\begin{aligned}
 &= (2 * f_a * e) / (D - e) \\
 &= (2*159996.34*13.89)/(762.00-13.89) \\
 &= 5.94 \text{ N/mm}^2
 \end{aligned}$$

Actual Stress at Given Thickness:

$$\begin{aligned}
 &= (P * Ri) / e_{cor} \\
 &= (1.00*370.28)/10.7156 \\
 &= 34553.55 \text{ kPa}
 \end{aligned}$$

As the Minimum Design Temperature is: 50.0 °C , Impact is Not Required

per Annex D

Cylindrical Shell From 40 To 50 A-106 B at -40 °C

Main Shell

Thickness Due to Internal Pressure (er):

$$\begin{aligned}
 &= (P * Do) / (2 * f + p) \text{ per 3.5.1.2 Eqn. 3.5.1-2} \\
 &= (1.30*762.0000)/(2*159996.34+1.30) \\
 &= 3.0830 + 3.1750 = 6.2580 \text{ mm}
 \end{aligned}$$

Max. All. Working Pressure at Given Thickness (MAWP):

$$= (2 * f * e_{cor}) / (D_{oca} - e_{cor})$$

$$= (2*159996.34*10.72)/(762.00-10.72)$$

$$= 4.56 \text{ N/mm}^2$$

Maximum Allowable Pressure, New and Cold (MAPNC):

$$= (2 * f_a * e) / (D - e)$$

$$= (2*159996.34*13.89)/(762.00-13.89)$$

$$= 5.94 \text{ N/mm}^2$$

Actual Stress at Given Thickness:

$$= (P * R_i) / e_{cor}$$

$$= (1.30*370.28)/10.7156$$

$$= 44919.61 \text{ kPa}$$

As the Minimum Design Temperature is: 50.0 °C , Impact is Not Required

per Annex D

Cylindrical Shell From 60 To 70 A-106 B at -40 °C

Right Channel

Thickness Due to Internal Pressure (e_r):

$$= (P * D_o) / (2 * f + p) \text{ per 3.5.1.2 Eqn. 3.5.1-2}$$

$$= (1.00*762.0000)/(2*159996.34+1.00)$$

$$= 2.3737 + 3.1750 = 5.5487 \text{ mm}$$

Max. All. Working Pressure at Given Thickness (MAWP):

$$= (2 * f * e_{cor}) / (D_{oca} - e_{cor})$$

$$= (2*159996.34*10.72)/(762.00-10.72)$$

$$= 4.56 \text{ N/mm}^2$$

Maximum Allowable Pressure, New and Cold (MAPNC):

$$= (2 * f_a * e) / (D - e)$$

$$= (2*159996.34*13.89)/(762.00-13.89)$$

$$= 5.94 \text{ N/mm}^2$$

Actual Stress at Given Thickness:

$$= (P * R_i) / e_{cor}$$

$$= (1.00 * 370.28) / 10.7156$$

$$= 34553.55 \text{ kPa}$$

As the Minimum Design Temperature is: 50.0 °C , Impact is Not Required

per Annex D

Welded Flat Head From 70 To 80 P355NH at -40 °C

Right head

Thickness Due to Internal Pressure (er):

$$= C * D * \sqrt{p/f} \text{ per 3.5.5-1}$$

$$= 0.4100 * 739.35 * \sqrt{1.00 / 191484.98}$$

$$= 21.9055 + 3.1750 = 25.0805 \text{ mm}$$

Max. All. Working Pressure at Given Thickness (MAWP):

$$= f * (e_{cor} / (C * D_{ca}))^2$$

$$= 191484.98 * (26.83 / (0.41 * 739.35))^2$$

$$= 1.88 \text{ N/mm}^2$$

Maximum Allowable Pressure, New and Cold (MAPNC):

$$= f_a * (e / (C * D))^2$$

$$= 191484.98 * (30.00 / (0.41 * 733.00))^2$$

$$= 1.91 \text{ N/mm}^2$$

Actual Stress at Given Thickness:

$$= 102093.97 \text{ kPa}$$

Note : For flat heads with openings, C should be the Maximum of CY1 and

0.41*Y2 per 3.5.5-5. d/D <= 0.5.

NOTE: If there are nozzles in the Flat Head,

check the Nozzle Summary for Final Flat Head Thickness
requirement.

As the Minimum Design Temperature is: 50.0 °C , Impact is Not Required
per Annex D

Hydrostatic Test Pressure Results:

PD:5500 5.8.5-1, Standard Test Pressure, Shell Side:

$$\begin{aligned}
 &= 1.25 * P * f_a / f_t * (t / (t - c)) \text{ (per 5.8.5-1)} \\
 &= 1.25 * 1.3000 * 1.000 * (13.891 / (13.891 - 3.175)) \\
 &= 2.1065 \text{ N/mm}^2
 \end{aligned}$$

PD:5500 5.8.5-1, Field Test Pressure, Shell Side:

$$\begin{aligned}
 &= 1.25 * P * f_a / f_t \text{ (per 5.8.5-1)} \\
 &= 1.25 * 1.3000 * 159996.344 / 159996.344 \\
 &= 1.6250 \text{ N/mm}^2
 \end{aligned}$$

Lower Bound Test Pressure, Shell Side	2.106	N/mm ²
Upper Bound Test Pressure, Shell Side	2.106	N/mm ²
1.35 times the Specified Design P, Shell Side	1.755	N/mm ²
Lower Bound Test Pressure, Channel Side	1.327	N/mm ²
Upper Bound Test Pressure, Channel Side	1.620	N/mm ²
1.35 times the Specified Design P, Channel Side	1.350	N/mm ²

Stresses on Elements due to Test Pressure:

From To	Stress	Allowable (90% of Yield)	Ratio

Left Head	136563.4	274494.7	0.498
Left Channel Shell	35873.9	215994.0	0.166
Main Shell	55865.5	215994.0	0.259

FileName : Gatti 750 heat exchanger

Internal Pressure Calculations : Step: 5 10:06a Dec 18, 2019

Right Channel	35873.9	215994.0	0.166
Right head	136190.6	274494.7	0.496

[Elements Suitable for Internal Pressure.](#)

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External Pressure Calculation Results :

British Standard PD 5500:2018

Welded Flat Head

Left Head

Note: This element's required thickness was computed in the internal Pressure Report using the maximum of the Internal and External pressures.

Cylindrical Shell From 20 to 30

Left Channel Shell

Determine the Stress Yield point [Py]:

$$\begin{aligned}
 &= S_{fact} * f * e_a / R \\
 &= 1.4 * 159996.344 * 10.716 / 375.64 \\
 &= 6389.711 \text{ kPa}
 \end{aligned}$$

Strain Factor [Z]:

$$\begin{aligned}
 &= p_i * R / L \\
 &= 3.141 * 375.642 / 250.00 \\
 &= 4.720
 \end{aligned}$$

Determine the ratio Pm/Py:

$$\begin{aligned}
 &= P_m / P_y \\
 &= 46786.727 / 6389.711 \\
 &= 7.000
 \end{aligned}$$

Determine delta from Figure 3.6-4:

$$= 0.639$$

Determine the critical Strain [strain]:

$$= 1/(n^2-1+Z^2) [1/(n^2/Z^2+1)^2 + ea^2/(12R^2(.91)) * (n^2-1+Z^2)^2]$$

$$= 0.0078763$$

Where n is the expected number of lobes [n]:

$$= 7$$

Determine the Elastic Instability Pressure [Pm]:

$$= E * ea * strain / R$$

$$= 208250.0 * 10.716 * 0.007876/375.64$$

$$= 46.789 \text{ N/mm}^2$$

External Allowable Pressure [Pmax]:

$$= \delta * Py$$

$$= 0.639 * 6.390$$

$$= 4.083 \text{ N/mm}^2$$

Cylindrical Shell From 40 to 50

Main Shell

Determine the Stress Yield point [Py]:

$$= S_{fact} * f * ea / R$$

$$= 1.4 * 159996.344 * 10.716/375.64$$

$$= 6389.711 \text{ kPa}$$

Strain Factor [Z]:

$$= \pi * R / L$$

$$= 3.141 * 375.642/4840.00$$

$$= 0.244$$

Determine the ratio Pm/Py:

$$= P_m / P_y$$

$$= 1803.113/6389.711$$

$$= 0.282$$

Determine delta from Figure 3.6-4:

$$= 0.094$$

Determine the critical Strain [strain]:

$$= 1/(n^2-1+Z^2) [1/(n^2/Z^2+1)^2 + ea^2/(12R^2(.91)) * (n^2-1+Z^2)^2]$$

$$= 0.0003010$$

Where n is the expected number of lobes [n]:

$$= 2$$

Determine the Elastic Instability Pressure [Pm]:

$$= E * ea * strain / R$$

$$= 210000.0 * 10.716 * 0.000301 / 375.64$$

$$= 1.803 \text{ N/mm}^2$$

External Allowable Pressure [Pmax]:

$$= \delta * Py$$

$$= 0.094 * 6.390$$

$$= 0.599 \text{ N/mm}^2$$

Cylindrical Shell From 60 to 70

Right Channel

Determine the Stress Yield point [Py]:

$$= S_{fact} * f * ea / R$$

$$= 1.4 * 159996.344 * 10.716 / 375.64$$

$$= 6389.711 \text{ kPa}$$

Strain Factor [Z]:

$$= \pi * R / L$$

$$= 3.141 * 375.642 / 250.00$$

$$= 4.720$$

Determine the ratio Pm/Py:

$$= P_m / P_y$$

$$= 47179.891/6389.711$$

$$= 7.000$$

Determine delta from Figure 3.6-4:

$$= 0.639$$

Determine the critical Strain [strain]:

$$= 1/(n^2-1+Z^2) [1/(n^2/Z^2+1)^2 + ea^2/(12R^2(.91)) * (n^2-1+Z^2)^2]$$

$$= 0.0078763$$

Where n is the expected number of lobes [n]:

$$= 7$$

Determine the Elastic Instability Pressure [Pm]:

$$= E * ea * strain / R$$

$$= 210000.0 * 10.716 * 0.007876/375.64$$

$$= 47.183 \text{ N/mm}^2$$

External Allowable Pressure [Pmax]:

$$= \delta * P_y$$

$$= 0.639 * 6.390$$

$$= 4.083 \text{ N/mm}^2$$

Welded Flat Head

Right head

Note: This element's required thickness was computed in the internal Pressure Report using the maximum of the Internal and External pressures.

External Pressure Calculations

	Section	Outside	Corroded	Factor	Factor
--	---------	---------	----------	--------	--------

FileName : Gatti 750 heat exchanger

External Pressure Calculations : Step: 6 10:06a Dec 18,2019

From	To	Length	Diameter	Thickness	A	B
		m	mm	mm		kPa

10	20	No Calc	...	26.8250	No Calc	No Calc
20	30	0.25000	762.000	10.7156	No Calc	No Calc
30	40	No Calc	...	51.8250	No Calc	No Calc
40	50	4.84000	762.000	10.7156	No Calc	No Calc
50	60	No Calc	...	51.8250	No Calc	No Calc
60	70	0.25000	762.000	10.7156	No Calc	No Calc
70	80	No Calc	...	26.8250	No Calc	No Calc

External Pressure Calculations

		External	External	External	External
From	To	Actual T.	Required T.	Des. Press.	M.A.W.P.
		mm	mm	N/mm ²	N/mm ²

10	20	30.0000	No Calc	0.10300	No Calc
20	30	13.8906	4.66488	0.10300	4.08326
30	40	55.0000	46.5328	0.10300	No Calc
40	50	13.8906	7.95622	0.10300	0.59949
50	60	55.0000	46.5328	0.10300	No Calc
60	70	13.8906	4.66008	0.10300	4.08326
70	80	30.0000	No Calc	0.10300	No Calc
Minimum				0.599	

External Pressure Calculations

		Actual Len.	Allow. Len.	Ring Inertia	Ring Inertia
From	To	Bet. Stiff.	Bet. Stiff.	Required	Available
		m	m	mm**4	mm**4

10	20	No Calc	No Calc	No Calc	No Calc
20	30	0.25000	17.89E+09	No Calc	No Calc
30	40	No Calc	No Calc	No Calc	No Calc
40	50	4.84000	173.0E+09	No Calc	No Calc

PV Elite 2018 Licensee: XGM
FileName : Gatti 750 heat exchanger
External Pressure Calculations :

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50	60	No Calc	No Calc	No Calc	No Calc
60	70	0.25000	19.23E+09	No Calc	No Calc
70	80	No Calc	No Calc	No Calc	No Calc

[Elements Suitable for External Pressure.](#)

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Element and Detail Weights

From	To	Element Metal Wgt.	Element ID Volume	Corroded Metal Wgt.	Corroded ID Volume	Extra due Misc %
		kgm	m ³	kgm	m ³	kgm

10	20	99.1429	...	88.6503	...	...
20	30	63.7685	0.10587	49.4073	0.10771	...
30	40	48.0423	...	48.0423	...	...
40	50	1234.56	1.47275	956.525	1.50836	...
50	60	48.0423	...	48.0423	...	...
60	70	63.7685	0.10587	49.4073	0.10771	...
70	80	98.8730	...	88.4089	...	...

Total		1656	1.68	1328	1.72	0

For elements specified as shell side elements, the volume(s) shown
above for those elements, reflects the displacement of the tubes.

Weight of Details

From	Type	Weight of Detail	X Offset, Dtl. Cent.	Y Offset, Dtl. Cent.	Description
		kgm	m	m	

10	Noz1	12.5500	2.40000	0.41302	Noz N1 Fr10
10	Noz1	47.0979	2.40000	0.41302	Noz N2 Fr10
20	Noz1	1.02537	0.14000	0.40521	Noz N1 Fr20
40	Sd1	20.2128	4.20000	0.45761	New Sd1
40	Sd1	20.2128	0.70000	0.45761	Sd1 2 Fr40
40	Noz1	28.5333	0.20000	0.43696	Noz N1 Fr40
40	Noz1	0.24032	0.45000	0.38616	Noz N2 Fr40
40	Noz1	1.02537	2.50000	0.40521	Noz N3 Fr40
40	Noz1	0.53155	4.55000	0.39568	Noz N4 Fr40
30	FTsh	126.044	0.078000	...	Gatti

FileName : Gatti 750 heat exchanger

Element and Detail Weights :

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30 Tube	1650.38	2.51600	...
30 RTsh	126.044	4.95400	...

Total Weight of Each Detail Type

Total Weight of Saddles	40.4
-------------------------	------

Total Weight of Nozzles	91.0
-------------------------	------

Total Weight of Exchanger Components	1902.5
--------------------------------------	--------

Sum of the Detail Weights	2033.9 kgm
---------------------------	------------

Weight Summation

Fabricated	Shop Test	Shipping	Erected	Empty	Operating
1656.2	3690.1	1656.2	3690.1	1656.2	3690.1
40.4	1683.5	40.4	...	40.4	...
91.0	...	91.0	...	...	...
...	372.2	...	...	...	...
...	...	...	...	...	...
...	...	...	...	...	...
...	...	...	...	91.0	...
1902.5	...	1902.5	...	...	...
...	...	...	...	1902.5	...
3690.1	5745.7	3690.1	3690.1	3690.1	3690.1 kgm

Note: The shipping total has been modified because some items have been specified as being installed in the shop.

Weight Summary

Fabricated Wt.	- Bare Weight w/O Removable Internals	3690.1 kgm
Shop Test Wt.	- Fabricated Weight + Water (Full)	5745.7 kgm

Shipping Wt.	- Fab. Wt + Rem. Intls.+ Shipping App.	3690.1 kgm
Erected Wt.	- Fab. Wt + Rem. Intls.+ Insul. (etc)	3690.1 kgm
Ope. Wt. no Liq	- Fab. Wt + Intls. + Details + Wghts.	3690.1 kgm
Operating Wt.	- Empty Wt + Operating Liq. Uncorroded	3690.1 kgm
Oper. Wt. + CA	- Corr Wt. + Operating Liquid	3362.4 kgm
Field Test Wt.	- Empty Weight + Water (Full)	5745.7 kgm

Exchanger Tube Data

Volume of Exchanger tubes :	0.4 m ³
Weight of Ope Liq in tubes :	0.0 kgm
Weight of Water in tubes :	372.2 kgm

Note: The Corroded Weight and thickness are used in the Horizontal

Vessel Analysis (Ope Case) and Earthquake Load Calculations.

Outside Surface Areas of Elements

		Surface	
From	To	Area	
		m ²	

10	20	...	
20	30	0.59833	
30	40	0.25822	
40	50	11.5837	
50	60	0.25822	
60	70	0.59833	
70	80	...	

Total		13.297 m ²	

Nozzle Flange MAWP Results :

Nozzle	----- Flange Rating				
Description	Operating	Ambient	Temperature	Class	Grade Group
	N/mm ²	N/mm ²	°C		

Noz N1 Fr10	2.0	2.0	-40	150	GR 1.1
Noz N2 Fr10	2.0	2.0	-40	150	GR 1.1

Shellside Flange Rating

Channelside Flange Rating

Lowest Flange Pressure Rating was (Ope)[TubeSide]: 1.960 N/mm²Lowest Flange Pressure Rating was (Amb)[TubeSide]: 1.960 N/mm²

Note: ANSI Ratings are per ANSI/ASME B16.5 2009 Metric Edition

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Wind Analysis Results

User Entered Importance Factor is	1.000	
Gust Factor (Gh, Gbar) Static Dynamic	1.381	
Shape Factor (Cf) for the Vessel is	0.601	
User Entered Basic Wind Speed	31.3	m/sec
Exposure Category	C	
Table Lookup Value Alpha from Table C6	7.0000	
Table Lookup Value Zg from Table C6	900.0000	
Table Lookup Value Do from Table C6	0.0050	

Wind Load Results per ASCE-7 93:

Sample Calculation for the First Element:

Roughness Factor = 1.000

Values [cf1] and [cf2]

Because RoughFact = 1 and DQZ > 2.5 and H/D > 7.0

Interpolating to find the final cf:

Because H / D < 25.0

$$\begin{aligned} CF &= CF1 + (CF2 - CF1) * (H/D - 7.0) / (25.0 - 7.0) \\ &= 0.600 + (0.700 - 0.600) * (7.202 - 7.0) / (25.0 - 7.0) \\ &= 0.601 \end{aligned}$$

Value of Alpha, Zg is taken from Table C6-2 [Alpha, Zg]

For Exposure Category C:

Alpha = 7.000 , Zg = 274.320 m

Height of Interest for First Element [z]

= Centroid Hgt + Base Height

= 0.572 + 0.000 = 0.572 m

but: z = Max(4.572 , 0.572) = 4.572 m

Note: Because z < 15 feet, use 15 feet to compute kz.

Velocity Pressure Coefficient [kZ]:

$$= 2.58 \left(\frac{z}{z_g} \right)^{2/\alpha} : z \text{ is Elevation of First Element}$$

$$= 2.58 \left(\frac{4.572}{900} \right)^{2/7.0}$$

$$= 0.801$$

Determine if Static or Dynamic Gust Factor Applies

Height to Diameter ratio :

$$= \text{Maximum Height}(\text{length})^2 / \text{Sum of Area of the Elements}$$

$$= 17.953 \text{ (}^2\text{)}/44.755$$

$$= 7.202$$

Vibration Frequency = 33.000 Hz

Because H/D > 5 Or Frequency < 1.0: Dynamic Analysis Implemented

$$\text{Element O/Dia} = 3 \text{ m}$$

$$\text{Vibration Damping Factor (Operating) Beta} = 0.01000$$

For Terrain Category C

$$S = 1.000, \text{ Gamma} = 0.230, \text{ Drag Coeff.} = 0.005, \text{ Alpha} = 7.000$$

Compute [fbar]

$$= 10.5 * \text{Frequency(Hz)} * \text{Vessel Height(ft)} / (S * \text{Vr(mph)})$$

$$= 10.5 * 33.000 \text{ (Hz)} * 17.953 \text{ (ft)}/S * 1.000 \text{ (mph)}$$

$$= 88.866$$

$$\text{Because FBAR} > 40: \text{FBAR} = 40.000$$

Wind Pressure - (performed in Imperial Units) [qz]

$$\text{Importance Factor: I} = 1.000$$

$$\text{Wind Speed} = 31.292 \text{ m/sec Converts to } 70.000 \text{ mph}$$

$$qz = 0.00256 * kZ * (I * \text{Vr})^2$$

$$= 0.00256 * 0.801 * (1.000 * 70.000)^2 = 10.046 \text{ psf}$$

$$\text{Converts to: } 0.481 \text{ kPa}$$

Force on the First Element [Fz]

$$= qz * Gh * CF * \text{Wind Area}$$

FileName : Gatti 750 heat exchanger

Wind Load Calculation :

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$$= 0.481 * 1.381 * 0.601 * 0.029$$

$$= 11.416 \text{ N}$$

Element	z	GH	Area	qz	Force
	m		m ²	kPa	N

Left Head	0.6	1.381	0.0	0.5	11.4
Left Channel Sh	0.6	1.381	0.2	0.5	91.3
Flange 30	0.6	1.381	0.0	0.5	0.0
Main Shell	0.6	1.381	4.4	0.5	1767.6
Flange 50	0.6	1.381	0.0	0.5	19.7
Right Channel	0.6	1.381	0.2	0.5	91.3
Right head	0.6	1.381	0.0	0.5	11.4

Wind Load Calculation

From	To	Wind Height	Wind Diameter	Wind Area	Wind Pressure	Element Wind Load
		m	m	m ²	kPa	N

10	20	0.57200	0.95280	0.028577	0.48101	11.4163
20	30	0.57200	0.91440	0.22854	0.48101	91.3013
30	40	0.57200	...	...	0.48101	...
40	50	0.57200	0.91440	4.42460	0.48101	1767.59
50	60	0.57200	0.89776	0.049364	0.48101	19.7207
60	70	0.57200	0.91440	0.22854	0.48101	91.3013
70	80	0.57200	0.95160	0.028541	0.48101	11.4019

Earthquake Analysis Results

The UBC Zone Factor for the Vessel is 0.0000
 The Importance Factor as Specified by the User is . 1.000
 The UBC Frequency and Soil Factor (C) is 2.750
 The UBC Force Factor as Specified by the User is .. 3.000
 The UBC Total Weight (W) for the Vessel is 32971.5 N
 The UBC Total Shear (V) for the Vessel is 0.0 N
 The UBC Top Shear (Ft) for the Vessel is 0.0 N

Earthquake Load Calculation

		Earthquake	Earthquake	Element	
From	To	Height	Weight	Ope	Load
		m	N	N	

10	20	0.36700	3663.50	...	
20	30	0.36711	3663.50	...	
30	40	...	3663.50	...	
40	Sad1	0.36711	3663.50	...	
Sad1	50	0.36711	3663.50	...	
40	50	0.36711	3663.50	...	
50	60	0.37407	3663.50	...	
60	70	0.36711	3663.50	...	
70	80	0.36650	3663.50	...	

Lifting Lug Calculations: Lug(s) on Left End of Vessel

Input Values:

Lifting Lug Material		P355GH
Lifting Lug Yield Stress	Yield	354991.88 kPa
Total Height of Lifting Lug	w	180.0000 mm
Thickness of Lifting Lug	t	12.0000 mm
Diameter of Hole in Lifting Lug	dh	60.0000 mm
Radius of Semi-Circular Arc of Lifting Lug	r	60.0000 mm
Height of Lug from bottom to Center of Hole	h	90.0000 mm
Offset from Vessel OD to Center of Hole	off	90.0000 mm
Lug Fillet Weld Size	tw	8.0000 mm
Length of weld along side of Lifting Lug	wl	190.0000 mm
Length of Weld along Bottom of Lifting Lug	wb	16.0000 mm
Thickness of Collar (if any)	tc	0.0000 mm
Diameter of Collar (if any)	dc	760.0000 mm
Impact Factor	Impfac	1.50
Sling Angle from Horizontal		90.0000 deg
Number of Lugs in Group		1

Lifting Lug Orientation to Vessel: Perpendicular

Lift Orientation : Horizontal Lift

PV Elite does not compute weak axis bending forces on the lugs. It is assumed that a spreader bar is used.

Computed Results:

Force Along Vessel Axis	Fax	0.00 N
Force Normal to Vessel	Fn	26654.40 N
Force Tangential to Vessel	Ft	0.00 N

Converting the weld leg dimension (tw) to the weld throat dimension.

Weld Group Inertia Calculations:

Weld Group Inertia about the Circumferential Axis	I _{lc}	8198377.000 mm**4
Weld Group Centroid distance in the Long. Direction	Y _{ll}	100.656 mm
Dist. of Weld Group Centroid from Lug bottom	Y _{ll_b}	95.000 mm
Weld Group Inertia about the Longitudinal Axis	I _{ll}	31215.977 mm**4
Weld Group Centroid Distance in the Circ. Direction	Y _{lc}	6.000 mm

Note: The Impact Factor is applied to the Forces acting on the Lug.

Primary Shear Stress in the Welds due to Shear Loads [S_{sl}]:

$$\begin{aligned}
 &= \sqrt{F_{ax}^2 + F_t^2 + F_n^2} / ((2 * (w_l + w_b)) * t_w) \\
 &= \sqrt{0^2 + 0^2 + 26654^2} / ((2 * (190.0 + 16.0)) * 5.6560) \\
 &= 11438.63 \text{ kPa}
 \end{aligned}$$

Shear Stress in the Welds due to Bending Loads [S_{blf}]:

$$\begin{aligned}
 &= (F_n * (h - Y_{ll_b})) * Y_{ll} / I_{lc} + (F_{ax} * off * Y_{ll} / I_{lc}) + (F_t * off * Y_{lc} / I_{ll}) \\
 &= (26654 * (90.000 - 95.000)) * 100.656 / 8198377 + \\
 &\quad (0 * 90.000 * 100.656 / 8198377) + \\
 &\quad (0 * 90.000 * 6.000 / 31215.977) \\
 &= -1636.30 \text{ kPa}
 \end{aligned}$$

Total Shear Stress for Combined Loads [S_t]:

$$\begin{aligned}
 &= S_{sl} + S_{blf} \\
 &= 11438.630 + -1636.300 \\
 &= 9802.33 \text{ kPa}
 \end{aligned}$$

Allowable Shear Stress for Combined Loads [S_a]:

$$\begin{aligned}
 &= 0.4 * Yield * Occfac \text{ (AISC Shear Allowable)} \\
 &= 0.4 * 354991 * 1.00 \\
 &= 141996.75 \text{ kPa}
 \end{aligned}$$

Shear Stress in Lug above Hole [S_{hs}]:

$$\begin{aligned}
 &= \sqrt{P_l^2 + F_{ax}^2} / S_{ha} \\
 &= \sqrt{26654^2 + 0^2} / 0.001
 \end{aligned}$$

$$= 37021.00 \text{ kPa}$$

Allowable Shear Stress in Lug above Hole [Sas]:

$$= 0.4 * \text{Yield} * \text{Occfac}$$

$$= 0.4 * 354991 * 1.00$$

$$= 141996.75 \text{ kPa}$$

Pin Hole Bearing Stress [Pbs]:

$$= \sqrt{F_{ax}^2 + F_n^2} / (t * d_h)$$

$$= \sqrt{0^2 + 26654^2} / (12.000 * 60.000)$$

$$= 37021.00 \text{ kPa}$$

Allowable Bearing Stress [Pba]:

$$= \min(0.75 * \text{Yield} * \text{Occfac}, 0.9 * \text{Yield}) \text{ AISC Bearing All.}$$

$$= \min(0.75 * 354991 * 1.00, 319492.7)$$

$$= 266243.91 \text{ kPa}$$

Bending Stress at the Base of the Lug [Fbs]:

$$= F_t * \text{off} / (w * t^2 / 6) + F_{ax} * \text{off} / (w^2 * t / 6)$$

$$= 0 * 90.000 / (180.000 * 12.000^2 / 6) +$$

$$0 * 90.000 / (180.000^2 * 12.000 / 6)$$

$$= 0.00 \text{ kPa}$$

Tensile Stress at the Base of the Lug [Fa]:

$$= F_n / (w * t)$$

$$= 0 / (180.000 * 12.000)$$

$$= 12340.33 \text{ kPa}$$

Total Combined Stress at the Base of the Lug:

$$= F_{bs} + F_a$$

$$= 0.0 + 12340.3$$

$$= 12340.33 \text{ kPa}$$

Lug Allowable Stress for Bending and Tension:

$$= \min(0.66 * \text{Yield} * \text{Occfac}, 0.75 * \text{Yield})$$

$$= \min(0.66 * 354991 * 1.00, 266243.9)$$

= 234294.64 kPa

Required Shackle Pin Diameter [Spd]:

= $\sqrt{(2 * \sqrt{F_n^2 + F_{ax}^2}) / (\pi * \sigma_a)}$

= $\sqrt{2 * \sqrt{26654^2 + 0^2} / (\pi * 141996)}$

= 10.9318 mm

**WRC 107/537 Stress Analysis for the Lifting Lug to Shell Junction in
 the new and Cold Condition (no corrosion applied).**

*Note: Since $\beta_1/\beta_2 \geq 0.25$, C22 (C22p) is adjusted per table 6
 in paragraph 4.3 of WRC Bulletin 107.*

Input Echo, WRC107/537 Item 1, Description: Lift Lug

Diameter Basis for Vessel	Vbasis	ID
Cylindrical or Spherical Vessel	Cylsph	Cylindrical
Internal Corrosion Allowance	Cas	0.0000 mm
Vessel Diameter	Dv	734.219 mm
Vessel Thickness	Tv	13.891 mm
Design Temperature		37.78 °C
Attachment Type	Type	Rectangular
Parameter C11	C11	16.00 mm
Parameter C22	C22	64.00 mm
Thickness of Reinforcing Pad	Tpad	8.000 mm
Pad Parameter C11P	C11p	50.000 mm
Pad Parameter C22P	C22p	200.000 mm
Design Internal Pressure	Dp	0.000 N/mm ²
Include Pressure Thrust		No

External Forces and Moments in WRC 107/537 Convention:

Radial Load	(SUS)	P	-26654.4 N
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Longitudinal Shear	(SUS)	Vl	0.0	N
Circumferential Shear	(SUS)	Vc	0.0	N
Circumferential Moment	(SUS)	Mc	0.0	N-m
Longitudinal Moment	(SUS)	Ml	0.0	N-m
Torsional Moment	(SUS)	Mt	0.0	N-m

Use Interactive Control No

WRC107 Version Version March 1979

Include Pressure Stress Indices per Div. 2 No

Compute Pressure Stress per WRC-368 No

WRC 107 Stress Calculation for SUSstained loads:

Radial Load	P	-26654.4	N
Circumferential Shear	VC	0.0	N
Longitudinal Shear	VL	0.0	N
Circumferential Moment	MC	0.0	N-m
Longitudinal Moment	ML	0.0	N-m
Torsional Moment	MT	0.0	N-m

Dimensionless Parameters used : Gamma = 17.27

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.063	4C	3.408	(A, B)
N(PHI) / (P/Rm)	0.063	3C	3.345	(C, D)
M(PHI) / (P)	0.037	2C1	0.219	(A, B)
M(PHI) / (P)	0.037	1C	0.252	(C, D)
N(PHI) / (MC/(Rm**2 * Beta))	0.034	3A	0.060	(A, B, C, D)
M(PHI) / (MC/(Rm * Beta))	0.044	1A	0.105	(A, B, C, D)
N(PHI) / (ML/(Rm**2 * Beta))	0.053	3B	0.612	(A, B, C, D)
M(PHI) / (ML/(Rm * Beta))	0.048	1B	0.063	(A, B, C, D)

FileName : Gatti 750 heat exchanger

Lifting Lug Calcs : Left Side

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N(x)	/ (P/Rm)	0.051	3C	3.427	(A, B)
N(x)	/ (P/Rm)	0.051	4C	3.455	(C, D)
M(x)	/ (P)	0.053	1C1	0.226	(A, B)
M(x)	/ (P)	0.053	2C	0.186	(C, D)
N(x)	/ (MC/(Rm**2 * Beta))	0.034	4A	0.092	(A, B, C, D)
M(x)	/ (MC/(Rm * Beta))	0.061	2A	0.063	(A, B, C, D)
N(x)	/ (ML/(Rm**2 * Beta))	0.053	4B	0.148	(A, B, C, D)
M(x)	/ (ML/(Rm * Beta))	0.066	2B	0.098	(A, B, C, D)

Stress Concentration Factors Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction

		Stress Values at							
Type of		(kPa)							
Stress	Load	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		10976	10976	10976	10976	10774	10774	10774	10774
Circ. Bend. P		73157	-73157	73157	-73157	84105	-84105	84105	-84105
Circ. Memb. MC		0	0	0	0	0	0	0	0
Circ. Bend. MC		0	0	0	0	0	0	0	0
Circ. Memb. ML		0	0	0	0	0	0	0	0
Circ. Bend. ML		0	0	0	0	0	0	0	0
Tot. Circ. Str.		84134	-62181	84134	-62181	94879	-73331	94879	-73331
Long. Memb. P		11038	11038	11038	11038	11128	11128	11128	11128
Long. Bend. P		75290	-75290	75290	-75290	62208	-62208	62208	-62208
Long. Memb. MC		0	0	0	0	0	0	0	0
Long. Bend. MC		0	0	0	0	0	0	0	0
Long. Memb. ML		0	0	0	0	0	0	0	0
Long. Bend. ML		0	0	0	0	0	0	0	0
Tot. Long. Str.		86329	-64251	86329	-64251	73336	-51080	73336	-51080

FileName : Gatti 750 heat exchanger

Lifting Lug Calcs : Left Side

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Shear VC	0	0	0	0	0	0	0	0
Shear VL	0	0	0	0	0	0	0	0
Shear MT	0	0	0	0	0	0	0	0
Tot. Shear	0	0	0	0	0	0	0	0

Str. Int.	86329	64251	86329	64251	94879	73331	94879	73331

Dimensionless Parameters used : Gamma = 26.93

Dimensionless Loads for Cylindrical Shells at Pad edge:

Curves read for 1979	Beta	Figure	Value	Location

N(PHI) / (P/Rm)	0.198	4C	4.010	(A, B)
N(PHI) / (P/Rm)	0.198	3C	2.820	(C, D)
M(PHI) / (P)	0.118	2C1	0.091	(A, B)
M(PHI) / (P)	0.118	1C	0.127	(C, D)
N(PHI) / (MC/(Rm**2 * Beta))	0.106	3A	0.735	(A, B, C, D)
M(PHI) / (MC/(Rm * Beta))	0.136	1A	0.094	(A, B, C, D)
N(PHI) / (ML/(Rm**2 * Beta))	0.168	3B	2.995	(A, B, C, D)
M(PHI) / (ML/(Rm * Beta))	0.149	1B	0.040	(A, B, C, D)
N(x) / (P/Rm)	0.160	3C	3.373	(A, B)
N(x) / (P/Rm)	0.160	4C	4.298	(C, D)
M(x) / (P)	0.167	1C1	0.097	(A, B)
M(x) / (P)	0.167	2C	0.060	(C, D)
N(x) / (MC/(Rm**2 * Beta))	0.106	4A	1.033	(A, B, C, D)
M(x) / (MC/(Rm * Beta))	0.187	2A	0.043	(A, B, C, D)
N(x) / (ML/(Rm**2 * Beta))	0.168	4B	0.971	(A, B, C, D)
M(x) / (ML/(Rm * Beta))	0.206	2B	0.049	(A, B, C, D)

Stress Concentration Factors Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad

		Stress Values at							
Type of		(kPa)							
Stress	Load	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		20572	20572	20572	20572	14465	14465	14465	14465
Circ. Bend. P		75598	-75598	75598	-75598	105416	-105416	105416	-105416
Circ. Memb. MC		0	0	0	0	0	0	0	0
Circ. Bend. MC		0	0	0	0	0	0	0	0
Circ. Memb. ML		0	0	0	0	0	0	0	0
Circ. Bend. ML		0	0	0	0	0	0	0	0
Tot. Circ. Str.		96170	-55026	96170	-55026	119881	-90950	119881	-90950
Long. Memb. P		17301	17301	17301	17301	22049	22049	22049	22049
Long. Bend. P		80229	-80229	80229	-80229	49900	-49900	49900	-49900
Long. Memb. MC		0	0	0	0	0	0	0	0
Long. Bend. MC		0	0	0	0	0	0	0	0
Long. Memb. ML		0	0	0	0	0	0	0	0
Long. Bend. ML		0	0	0	0	0	0	0	0
Tot. Long. Str.		97530	-62927	97530	-62927	71949	-27850	71949	-27850
Shear VC		0	0	0	0	0	0	0	0
Shear VL		0	0	0	0	0	0	0	0
Shear MT		0	0	0	0	0	0	0	0
Tot. Shear		0	0	0	0	0	0	0	0
Str. Int.		97530	62927	97530	62927	119881	90950	119881	90950

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction

Type of Stress Int.	Stress Values at (kPa)							
Location	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)	0	0	0	0	0	0	0	0
Circ. Pl (SUS)	10976	10976	10976	10976	10774	10774	10774	10774
Circ. Q (SUS)	73157	-73157	73157	-73157	84105	-84105	84105	-84105
Long. Pm (SUS)	0	0	0	0	0	0	0	0
Long. Pl (SUS)	11038	11038	11038	11038	11128	11128	11128	11128
Long. Q (SUS)	75290	-75290	75290	-75290	62208	-62208	62208	-62208
Shear Pm (SUS)	0	0	0	0	0	0	0	0
Shear Pl (SUS)	0	0	0	0	0	0	0	0
Shear Q (SUS)	0	0	0	0	0	0	0	0
Pm (SUS)	0	0	0	0	0	0	0	0
Pm+Pl (SUS)	11038	11038	11038	11038	11128	11128	11128	11128
Pm+Pl+Q (Total)	86329	64251	86329	64251	94879	73331	94879	73331
Type of Stress Int.	Max. S.I.	S.I. Allowable			Result			
	kPa							
Pm (SUS)	0	159996			Passed			
Pm+Pl (SUS)	11128	239994			Passed			
Pm+Pl+Q (TOTAL)	94879	479989			Passed			

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge

Type of Stress Int.	Stress Values at (kPa)							
Location	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)	0	0	0	0	0	0	0	0
Circ. Pl (SUS)	20572	20572	20572	20572	14465	14465	14465	14465
Circ. Q (SUS)	75598	-75598	75598	-75598	105416	-105416	105416	-105416
Long. Pm (SUS)	0	0	0	0	0	0	0	0
Long. Pl (SUS)	17301	17301	17301	17301	22049	22049	22049	22049
Long. Q (SUS)	80229	-80229	80229	-80229	49900	-49900	49900	-49900
Shear Pm (SUS)	0	0	0	0	0	0	0	0
Shear Pl (SUS)	0	0	0	0	0	0	0	0
Shear Q (SUS)	0	0	0	0	0	0	0	0
Pm (SUS)	0	0	0	0	0	0	0	0
Pm+Pl (SUS)	20572	20572	20572	20572	22049	22049	22049	22049
Pm+Pl+Q (Total)	97530	62927	97530	62927	119881	90950	119881	90950

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
	kPa		
Pm (SUS)	0	159996	Passed
Pm+Pl (SUS)	22049	239994	Passed
Pm+Pl+Q (TOTAL)	119881	479989	Passed

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FileName : Gatti 750 heat exchanger

Lifting Lug Calcs : Left Side

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Lifting Lug Calculations: Lug(s) on Right End of Vessel

Input Values:

Lifting Lug Material		P355GH
Lifting Lug Yield Stress	Yield	354991.88 kPa
Total Height of Lifting Lug	w	180.0000 mm
Thickness of Lifting Lug	t	12.0000 mm
Diameter of Hole in Lifting Lug	dh	60.0000 mm
Radius of Semi-Circular Arc of Lifting Lug	r	60.0000 mm
Height of Lug from bottom to Center of Hole	h	90.0000 mm
Offset from Vessel OD to Center of Hole	off	90.0000 mm
Lug Fillet Weld Size	tw	8.0000 mm
Length of weld along side of Lifting Lug	wl	190.0000 mm
Length of Weld along Bottom of Lifting Lug	wb	16.0000 mm
Thickness of Collar (if any)	tc	0.0000 mm
Diameter of Collar (if any)	dc	760.0000 mm
Impact Factor	Impfac	1.50
Sling Angle from Horizontal		90.0000 deg
Number of Lugs in Group		1

Lifting Lug Orientation to Vessel: Perpendicular

Lift Orientation : Horizontal Lift

PV Elite does not compute weak axis bending forces on the lugs. It is assumed that a spreader bar is used.

Computed Results:

Force Along Vessel Axis	Fax	0.00 N
Force Normal to Vessel	Fn	27623.14 N
Force Tangential to Vessel	Ft	0.00 N

Converting the weld leg dimension (tw) to the weld throat dimension.

Weld Group Inertia Calculations:

Weld Group Inertia about the Circumferential Axis	I _{lc}	8198377.000 mm**4
Weld Group Centroid distance in the Long. Direction	Y _{ll}	100.656 mm
Dist. of Weld Group Centroid from Lug bottom	Y _{ll_b}	95.000 mm
Weld Group Inertia about the Longitudinal Axis	I _{ll}	31215.977 mm**4
Weld Group Centroid Distance in the Circ. Direction	Y _{lc}	6.000 mm

Note: The Impact Factor is applied to the Forces acting on the Lug.

Primary Shear Stress in the Welds due to Shear Loads [S_{sl}]:

$$\begin{aligned}
 &= \sqrt{F_{ax}^2 + F_t^2 + F_n^2} / ((2 * (w_l + w_b)) * t_w) \\
 &= \sqrt{0^2 + 0^2 + 27623^2} / ((2 * (190.0 + 16.0)) * 5.6560) \\
 &= 11854.36 \text{ kPa}
 \end{aligned}$$

Shear Stress in the Welds due to Bending Loads [S_{blf}]:

$$\begin{aligned}
 &= (F_n * (h - Y_{ll_b})) * Y_{ll} / I_{lc} + (F_{ax} * off * Y_{ll} / I_{lc}) + (F_t * off * Y_{lc} / I_{ll}) \\
 &= (27623 * (90.000 - 95.000)) * 100.656 / 8198377 + \\
 &\quad (0 * 90.000 * 100.656 / 8198377) + \\
 &\quad (0 * 90.000 * 6.000 / 31215.977) \\
 &= -1695.77 \text{ kPa}
 \end{aligned}$$

Total Shear Stress for Combined Loads [S_t]:

$$\begin{aligned}
 &= S_{sl} + S_{blf} \\
 &= 11854.360 + -1695.770 \\
 &= 10158.59 \text{ kPa}
 \end{aligned}$$

Allowable Shear Stress for Combined Loads [S_a]:

$$\begin{aligned}
 &= 0.4 * \text{Yield} * \text{Occfac} \text{ (AISC Shear Allowable)} \\
 &= 0.4 * 354991 * 1.00 \\
 &= 141996.75 \text{ kPa}
 \end{aligned}$$

Shear Stress in Lug above Hole [S_{hs}]:

$$\begin{aligned}
 &= \sqrt{P_l^2 + F_{ax}^2} / S_{ha} \\
 &= \sqrt{27623^2 + 0^2} / 0.001
 \end{aligned}$$

$$= 38366.51 \text{ kPa}$$

Allowable Shear Stress in Lug above Hole [Sas]:

$$= 0.4 * \text{Yield} * \text{Occfac}$$

$$= 0.4 * 354991 * 1.00$$

$$= 141996.75 \text{ kPa}$$

Pin Hole Bearing Stress [Pbs]:

$$= \sqrt{F_{ax}^2 + F_n^2} / (t * d_h)$$

$$= \sqrt{0^2 + 27623^2} / (12.000 * 60.000)$$

$$= 38366.51 \text{ kPa}$$

Allowable Bearing Stress [Pba]:

$$= \min(0.75 * \text{Yield} * \text{Occfac}, 0.9 * \text{Yield}) \text{ AISC Bearing All.}$$

$$= \min(0.75 * 354991 * 1.00, 319492.7)$$

$$= 266243.91 \text{ kPa}$$

Bending Stress at the Base of the Lug [Fbs]:

$$= F_t * \text{off} / (w * t^2 / 6) + F_{ax} * \text{off} / (w^2 * t / 6)$$

$$= 0 * 90.000 / (180.000 * 12.000^2 / 6) +$$

$$0 * 90.000 / (180.000^2 * 12.000 / 6)$$

$$= 0.00 \text{ kPa}$$

Tensile Stress at the Base of the Lug [Fa]:

$$= F_n / (w * t)$$

$$= 0 / (180.000 * 12.000)$$

$$= 12788.84 \text{ kPa}$$

Total Combined Stress at the Base of the Lug:

$$= F_{bs} + F_a$$

$$= 0.0 + 12788.8$$

$$= 12788.84 \text{ kPa}$$

Lug Allowable Stress for Bending and Tension:

$$= \min(0.66 * \text{Yield} * \text{Occfac}, 0.75 * \text{Yield})$$

$$= \min(0.66 * 354991 * 1.00, 266243.9)$$

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Lifting Lug Calcs : Right Side

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= 234294.64 kPa

Required Shackle Pin Diameter [Spd]:

= $\sqrt{(2 * \sqrt{F_n^2 + F_{ax}^2}) / (\pi * \sigma_a)}$ = $\sqrt{2 * \sqrt{(27623^2 + 0^2)} / (\pi * 141996)}$

= 11.1287 mm

WRC 107/537 Stress Analysis for the Lifting Lug to Shell Junction in the new and Cold Condition (no corrosion applied).

Note: Since $\beta_1/\beta_2 \geq 0.25$, C22 (C22p) is adjusted per table 6 in paragraph 4.3 of WRC Bulletin 107.

Input Echo, WRC107/537 Item 1, Description: Lift Lug

Diameter Basis for Vessel	Vbasis	ID
Cylindrical or Spherical Vessel	Cylsph	Cylindrical
Internal Corrosion Allowance	Cas	0.0000 mm
Vessel Diameter	Dv	734.219 mm
Vessel Thickness	Tv	13.891 mm
Design Temperature		37.78 °C
Attachment Type	Type	Rectangular
Parameter C11	C11	16.00 mm
Parameter C22	C22	64.00 mm
Thickness of Reinforcing Pad	Tpad	8.000 mm
Pad Parameter C11P	C11p	50.000 mm
Pad Parameter C22P	C22p	200.000 mm
Design Internal Pressure	Dp	0.000 N/mm ²
Include Pressure Thrust		No

External Forces and Moments in WRC 107/537 Convention:

Radial Load	(SUS)	P	-27623.1 N
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Lifting Lug Calcs : Right Side

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Longitudinal Shear	(SUS)	Vl	0.0	N
Circumferential Shear	(SUS)	Vc	0.0	N
Circumferential Moment	(SUS)	Mc	0.0	N-m
Longitudinal Moment	(SUS)	Ml	0.0	N-m
Torsional Moment	(SUS)	Mt	0.0	N-m

Use Interactive Control No

WRC107 Version Version March 1979

Include Pressure Stress Indices per Div. 2 No

Compute Pressure Stress per WRC-368 No

WRC 107 Stress Calculation for SUSstained loads:

Radial Load	P	-27623.1	N
Circumferential Shear	VC	0.0	N
Longitudinal Shear	VL	0.0	N
Circumferential Moment	MC	0.0	N-m
Longitudinal Moment	ML	0.0	N-m
Torsional Moment	MT	0.0	N-m

Dimensionless Parameters used : Gamma = 17.27

Dimensionless Loads for Cylindrical Shells at Attachment Junction:

Curves read for 1979	Beta	Figure	Value	Location
N(PHI) / (P/Rm)	0.063	4C	3.408	(A, B)
N(PHI) / (P/Rm)	0.063	3C	3.345	(C, D)
M(PHI) / (P)	0.037	2C1	0.219	(A, B)
M(PHI) / (P)	0.037	1C	0.252	(C, D)
N(PHI) / (MC/(Rm**2 * Beta))	0.034	3A	0.060	(A, B, C, D)
M(PHI) / (MC/(Rm * Beta))	0.044	1A	0.105	(A, B, C, D)
N(PHI) / (ML/(Rm**2 * Beta))	0.053	3B	0.612	(A, B, C, D)
M(PHI) / (ML/(Rm * Beta))	0.048	1B	0.063	(A, B, C, D)

FileName : Gatti 750 heat exchanger

Lifting Lug Calcs : Right Side

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N(x)	/ (P/Rm)	0.051	3C	3.427	(A,B)
N(x)	/ (P/Rm)	0.051	4C	3.455	(C,D)
M(x)	/ (P)	0.053	1C1	0.226	(A,B)
M(x)	/ (P)	0.053	2C	0.186	(C,D)
N(x)	/ (MC/(Rm**2 * Beta))	0.034	4A	0.092	(A,B,C,D)
M(x)	/ (MC/(Rm * Beta))	0.061	2A	0.063	(A,B,C,D)
N(x)	/ (ML/(Rm**2 * Beta))	0.053	4B	0.148	(A,B,C,D)
M(x)	/ (ML/(Rm * Beta))	0.066	2B	0.098	(A,B,C,D)

Stress Concentration Factors Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Attachment Junction

		Stress Values at							
Type of		(kPa)							
Stress	Load	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		11375	11375	11375	11375	11165	11165	11165	11165
Circ. Bend. P		75816	-75816	75816	-75816	87162	-87162	87162	-87162
Circ. Memb. MC		0	0	0	0	0	0	0	0
Circ. Bend. MC		0	0	0	0	0	0	0	0
Circ. Memb. ML		0	0	0	0	0	0	0	0
Circ. Bend. ML		0	0	0	0	0	0	0	0
Tot. Circ. Str.		87191	-64441	87191	-64441	98328	-75996	98328	-75996
Long. Memb. P		11439	11439	11439	11439	11532	11532	11532	11532
Long. Bend. P		78026	-78026	78026	-78026	64469	-64469	64469	-64469
Long. Memb. MC		0	0	0	0	0	0	0	0
Long. Bend. MC		0	0	0	0	0	0	0	0
Long. Memb. ML		0	0	0	0	0	0	0	0
Long. Bend. ML		0	0	0	0	0	0	0	0
Tot. Long. Str.		89466	-66587	89466	-66587	76001	-52936	76001	-52936

FileName : Gatti 750 heat exchanger

Lifting Lug Calcs : Right Side

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Shear VC	0	0	0	0	0	0	0	0
Shear VL	0	0	0	0	0	0	0	0
Shear MT	0	0	0	0	0	0	0	0
Tot. Shear	0	0	0	0	0	0	0	0

Str. Int.	89466	66587	89466	66587	98328	75996	98328	75996

Dimensionless Parameters used : Gamma = 26.93

Dimensionless Loads for Cylindrical Shells at Pad edge:

Curves read for 1979	Beta	Figure	Value	Location

N(PHI) / (P/Rm)	0.198	4C	4.010	(A, B)
N(PHI) / (P/Rm)	0.198	3C	2.820	(C, D)
M(PHI) / (P)	0.118	2C1	0.091	(A, B)
M(PHI) / (P)	0.118	1C	0.127	(C, D)
N(PHI) / (MC/(Rm**2 * Beta))	0.106	3A	0.735	(A, B, C, D)
M(PHI) / (MC/(Rm * Beta))	0.136	1A	0.094	(A, B, C, D)
N(PHI) / (ML/(Rm**2 * Beta))	0.168	3B	2.995	(A, B, C, D)
M(PHI) / (ML/(Rm * Beta))	0.149	1B	0.040	(A, B, C, D)
N(x) / (P/Rm)	0.160	3C	3.373	(A, B)
N(x) / (P/Rm)	0.160	4C	4.298	(C, D)
M(x) / (P)	0.167	1C1	0.097	(A, B)
M(x) / (P)	0.167	2C	0.060	(C, D)
N(x) / (MC/(Rm**2 * Beta))	0.106	4A	1.033	(A, B, C, D)
M(x) / (MC/(Rm * Beta))	0.187	2A	0.043	(A, B, C, D)
N(x) / (ML/(Rm**2 * Beta))	0.168	4B	0.971	(A, B, C, D)
M(x) / (ML/(Rm * Beta))	0.206	2B	0.049	(A, B, C, D)

Stress Concentration Factors Kn = 1.00, Kb = 1.00

Stresses in the Vessel at the Edge of Reinforcing Pad

		Stress Values at							
Type of		(kPa)							
Stress	Load	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Memb. P		21319	21319	21319	21319	14990	14990	14990	14990
Circ. Bend. P		78346	-78346	78346	-78346	109247	-109247	109247	-109247
Circ. Memb. MC		0	0	0	0	0	0	0	0
Circ. Bend. MC		0	0	0	0	0	0	0	0
Circ. Memb. ML		0	0	0	0	0	0	0	0
Circ. Bend. ML		0	0	0	0	0	0	0	0
Tot. Circ. Str.		99666	-57026	99666	-57026	124238	-94256	124238	-94256
Long. Memb. P		17930	17930	17930	17930	22851	22851	22851	22851
Long. Bend. P		83144	-83144	83144	-83144	51713	-51713	51713	-51713
Long. Memb. MC		0	0	0	0	0	0	0	0
Long. Bend. MC		0	0	0	0	0	0	0	0
Long. Memb. ML		0	0	0	0	0	0	0	0
Long. Bend. ML		0	0	0	0	0	0	0	0
Tot. Long. Str.		101075	-65214	101075	-65214	74564	-28862	74564	-28862
Shear VC		0	0	0	0	0	0	0	0
Shear VL		0	0	0	0	0	0	0	0
Shear MT		0	0	0	0	0	0	0	0
Tot. Shear		0	0	0	0	0	0	0	0
Str. Int.		101075	65214	101075	65214	124238	94256	124238	94256

WRC 107/537 Stress Summations:

Vessel Stress Summation at Attachment Junction

Type of Stress Int.	Stress Values at (kPa)							
Location	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)	0	0	0	0	0	0	0	0
Circ. Pl (SUS)	11375	11375	11375	11375	11165	11165	11165	11165
Circ. Q (SUS)	75816	-75816	75816	-75816	87162	-87162	87162	-87162
Long. Pm (SUS)	0	0	0	0	0	0	0	0
Long. Pl (SUS)	11439	11439	11439	11439	11532	11532	11532	11532
Long. Q (SUS)	78026	-78026	78026	-78026	64469	-64469	64469	-64469
Shear Pm (SUS)	0	0	0	0	0	0	0	0
Shear Pl (SUS)	0	0	0	0	0	0	0	0
Shear Q (SUS)	0	0	0	0	0	0	0	0
Pm (SUS)	0	0	0	0	0	0	0	0
Pm+Pl (SUS)	11439	11439	11439	11439	11532	11532	11532	11532
Pm+Pl+Q (Total)	89466	66587	89466	66587	98328	75996	98328	75996

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
	kPa		
Pm (SUS)	0	159996	Passed
Pm+Pl (SUS)	11532	239994	Passed
Pm+Pl+Q (TOTAL)	98328	479989	Passed

WRC 107/537 Stress Summations:

Vessel Stress Summation at Reinforcing Pad Edge

Type of Stress Int.	Stress Values at (kPa)							
Location	Au	Al	Bu	Bl	Cu	Cl	Du	Dl
Circ. Pm (SUS)	0	0	0	0	0	0	0	0
Circ. Pl (SUS)	21319	21319	21319	21319	14990	14990	14990	14990
Circ. Q (SUS)	78346	-78346	78346	-78346	109247	-109247	109247	-109247
Long. Pm (SUS)	0	0	0	0	0	0	0	0
Long. Pl (SUS)	17930	17930	17930	17930	22851	22851	22851	22851
Long. Q (SUS)	83144	-83144	83144	-83144	51713	-51713	51713	-51713
Shear Pm (SUS)	0	0	0	0	0	0	0	0
Shear Pl (SUS)	0	0	0	0	0	0	0	0
Shear Q (SUS)	0	0	0	0	0	0	0	0
Pm (SUS)	0	0	0	0	0	0	0	0
Pm+Pl (SUS)	21319	21319	21319	21319	22851	22851	22851	22851
Pm+Pl+Q (Total)	101075	65214	101075	65214	124238	94256	124238	94256

Type of Stress Int.	Max. S.I.	S.I. Allowable	Result
	kPa		
Pm (SUS)	0	159996	Passed
Pm+Pl (SUS)	22851	239994	Passed
Pm+Pl+Q (TOTAL)	124238	479989	Passed

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Lifting Lug Calcs : Right Side Step: 13 10:06a Dec 18, 2019

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ASME Horizontal Vessel Analysis: Stresses for the Left Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Operating Case

Input and Calculated Values:

Vessel Mean Radius	Rm	375.64	mm
Stiffened Vessel Length per 4.15.6	L	4.84	m
Distance from Saddle to Vessel tangent	a	242.00	mm
Saddle Width	b	204.00	mm
Saddle Bearing Angle	theta	120.00	degrees
Wear Plate Width	b1	306.00	mm
Wear Plate Bearing Angle	theta1	132.00	degrees
Wear Plate Thickness	tr	10.0	mm
Wear Plate Allowable Stress	Sr	186662.33	kPa
Shell Allowable Stress used in Calculation		159996.34	kPa
Head Allowable Stress used in Calculation		153188.00	kPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		19020.60	N

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	18850.32	159996.34	kPa
Long. Stress at Bottom of Midspan	26719.26	159996.34	kPa
Long. Stress at Top of Saddles	22691.75	159996.34	kPa
Long. Stress at Bottom of Saddles	22836.36	159996.34	kPa
Tangential Shear in Shell	4978.86	127997.08	kPa
Circ. Stress at Horn of Saddle	2396.35	199995.42	kPa

Circ. Compressive Stress in Shell 445.42 159996.34 kPa

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$\begin{aligned}
 &= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E \\
 &= 3.00 * (1992.7/2 + 0) * 572.0000/650.6314 \\
 &= 2627.9 \text{ N}
 \end{aligned}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$\begin{aligned}
 &= \max(F_l, \text{Friction Load}, \text{Sum of X Forces}) * B / L_s \\
 &= \max(263.23 , 0.00 , 0) * 572.0000/3499.9995 \\
 &= 43.0 \text{ N}
 \end{aligned}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$\begin{aligned}
 &= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st}) \\
 &= 16392 + \max(43 , 2627 , 0 , 0) \\
 &= 19020.6 \text{ N}
 \end{aligned}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	19218.81	N
Transverse Shear Load Saddle	996.37	N
Longitudinal Shear Load Saddle	263.23	N

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0246	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0434
K7P = 0.0203			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right] \\
 &= -19020 \cdot 0.24 \left[1 - \left(1 - \frac{0.24}{4.84} + \frac{(0.376^2 - 0.000^2)}{(2 \cdot 0.24 \cdot 4.84)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 4.84)} \right) \right] \\
 &= 47.1 \text{ N-m}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= Q \cdot L / 4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) - 4a/L \\
 &= 19020 \cdot 4 / 4 \left(1 + 2 \frac{(0^2 - 0^2)}{(4^2)} \right) / \left(1 + \frac{(4 \cdot 0)}{3 \cdot 4} \right) - 4 \cdot 0 / 4 \\
 &= 18696.8 \text{ N-m}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) - 18696.8 / (\pi \cdot 375.6^2 \cdot 10.716) \\
 &= 18850.32 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) + 18696.8 / (\pi \cdot 375.6^2 \cdot 10.716) \\
 &= 26719.26 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) - 47.1 / (0.1066 \cdot \pi \cdot 375.6^2 \cdot 10.716) \\
 &= 22691.75 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) + 47.1 / (0.1923 \cdot \pi \cdot 375.6^2 \cdot 10.716) \\
 &= 22836.36 \text{ kPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L-2a)/(L+(4*h^2/3)) \\
 &= 19020 (4.84 - 2 * 0.24)/(4.84 + (4 * 0.00/3)) \\
 &= 17118.5 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
 &= K2 * T / (Rm * t) \\
 &= 1.1707 * 17118.54/(375.6422 * 10.7156) \\
 &= 4978.86 \text{ kPa}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 * \text{sqrt}(Rm * t) \\
 &= 0.78 * \text{sqrt}(375.642 * 10.716) \\
 &= 49.487 \text{ mm}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 * Q * k / (t * (b + X1 + X2)) \\
 &= -0.7603 * 19020 * 0.1/(10.716 * (204.00 + 49.49 + 49.49)) \\
 &= -445.42 \text{ kPa}
 \end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
 &= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a) \\
 &= \min(204.00 + 1.56 * \text{sqrt}(375.642 * 10.716), 2 * 0.242) \\
 &= 302.97 \text{ mm}
 \end{aligned}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$\begin{aligned}
 &= \min(Sr/S, 1) \\
 &= \min(186662.328/159996.344, 1) \\
 &= 1.0000
 \end{aligned}$$

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$\begin{aligned}
 &= -K5 * Q * k / (B1(t + eta * tr)) \\
 &= -0.7603 * 19020 * 0.1/(302.974 (10.716 + 1.000 * 10.000)) \\
 &= -230.41 \text{ kPa}
 \end{aligned}$$

Circ. Comp. Stress at Horn of Saddle, L>=8Rm (4.15.27) [sigma7,r]:

$$= -Q/(4(t+eta*tr)b1) - 3*K7*Q/(2(t+eta*tr)^2)$$

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$$= -19020 / (4(10.716 + 1.000 * 10.000) 302.974) -$$

$$3 * 0.025 * 19020 / (2(10.716 + 1.000 * 10.000)^2)$$

$$= -2396.35 \text{ kPa}$$

ASME Horizontal Vessel Analysis: Stresses for the Right Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Input and Calculated Values:

Vessel Mean Radius	Rm	375.64	mm
Stiffened Vessel Length per 4.15.6	L	4.84	m
Distance from Saddle to Vessel tangent	a	242.00	mm
Saddle Width	b	204.00	mm
Saddle Bearing Angle	theta	120.00	degrees
Wear Plate Width	b1	306.00	mm
Wear Plate Bearing Angle	theta1	132.00	degrees
Wear Plate Thickness	tr	10.0	mm
Wear Plate Allowable Stress	Sr	95145.48	kPa
Shell Allowable Stress used in Calculation		159996.34	kPa
Head Allowable Stress used in Calculation		153187.98	kPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Operating Case		19206.60	N

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	18811.84	159996.34	kPa
Long. Stress at Bottom of Midspan	26757.74	159996.34	kPa
Long. Stress at Top of Saddles	22690.84	159996.34	kPa
Long. Stress at Bottom of Saddles	22836.86	159996.34	kPa
Tangential Shear in Shell	5027.55	127997.08	kPa

Circ. Stress at Horn of Saddle	3508.86	199995.42	kPa
Circ. Compressive Stress in Shell	449.78	159996.34	kPa

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$\begin{aligned}
 &= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E \\
 &= 3.00 * (1992.7/2 + 0) * 572.0000/650.6314 \\
 &= 2627.9 \text{ N}
 \end{aligned}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$\begin{aligned}
 &= \max(F_l, \text{Friction Load}, \text{Sum of X Forces}) * B / L_s \\
 &= \max(263.23, 0.00, 0) * 572.0000/3499.9995 \\
 &= 43.0 \text{ N}
 \end{aligned}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$\begin{aligned}
 &= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st}) \\
 &= 16578 + \max(43, 2627, 0, 0) \\
 &= 19206.6 \text{ N}
 \end{aligned}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	19404.80	N
Transverse Shear Load Saddle	996.37	N
Longitudinal Shear Load Saddle	263.23	N

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, $k = 0.1$

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0246	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0434
K7P = 0.0203			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right] \\
 &= -19206 \cdot 0.24 \left[1 - \left(1 - \frac{0.24}{4.84} + \frac{(0.376^2 - 0.000^2)}{(2 \cdot 0.24 \cdot 4.84)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 4.84)} \right) \right] \\
 &= 47.6 \text{ N-m}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= Q \cdot L / 4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) - 4a/L \\
 &= 19206 \cdot 4 / 4 \left(1 + 2 \frac{(0^2 - 0^2)}{(4^2)} \right) / \left(1 + \frac{(4 \cdot 0)}{(3 \cdot 4)} \right) - 4 \cdot 0 / 4 \\
 &= 18879.6 \text{ N-m}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) - 18879.6 / (\pi \cdot 375.6^2 \cdot 10.716) \\
 &= 18811.84 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) + 18879.6 / (\pi \cdot 375.6^2 \cdot 10.716) \\
 &= 26757.74 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) - 47.6 / (0.1066 \cdot \pi \cdot 375.6^2 \cdot 10.716) \\
 &= 22690.84 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\
 &= 1.300 \cdot 375.642 / (2 \cdot 10.716) + 47.6 / (0.1923 \cdot \pi \cdot 375.6^2 \cdot 10.716) \\
 &= 22836.86 \text{ kPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L-2a)/(L+(4*h^2/3)) \\
 &= 19206 (4.84 - 2 * 0.24) / (4.84 + (4 * 0.00/3)) \\
 &= 17285.9 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
 &= K2 * T / (Rm * t) \\
 &= 1.1707 * 17285.94 / (375.6422 * 10.7156) \\
 &= 5027.55 \text{ kPa}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 * \text{sqrt}(Rm * t) \\
 &= 0.78 * \text{sqrt}(375.642 * 10.716) \\
 &= 49.487 \text{ mm}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 * Q * k / (t * (b + X1 + X2)) \\
 &= -0.7603 * 19206 * 0.1 / (10.716 * (204.00 + 49.49 + 49.49)) \\
 &= -449.78 \text{ kPa}
 \end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
 &= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a) \\
 &= \min(204.00 + 1.56 * \text{sqrt}(375.642 * 10.716), 2 * 0.242) \\
 &= 302.97 \text{ mm}
 \end{aligned}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$\begin{aligned}
 &= \min(Sr/S, 1) \\
 &= \min(95145.477/159996.344, 1) \\
 &= 0.5947
 \end{aligned}$$

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$\begin{aligned}
 &= -K5 * Q * k / (B1(t + eta * tr)) \\
 &= -0.7603 * 19206 * 0.1 / (302.974 (10.716 + 0.595 * 10.000)) \\
 &= -289.26 \text{ kPa}
 \end{aligned}$$

Circ. Comp. Stress at Horn of Saddle, L>=8Rm (4.15.27) [sigma7,r]:

$$= -Q/(4(t+\eta \cdot tr)b_1) - 3 \cdot K_7 \cdot Q/(2(t+\eta \cdot tr)^2)$$

$$= -19206/(4(10.716 + 0.595 \cdot 10.000)302.974) -$$

$$3 \cdot 0.025 \cdot 19206/(2(10.716 + 0.595 \cdot 10.000)^2)$$

$$= -3508.86 \text{ kPa}$$

ASME Horizontal Vessel Analysis: Stresses for the Left Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Horizontal Vessel Stress Calculations : Test Case

Input and Calculated Values:

Vessel Mean Radius	Rm	374.05	mm
Stiffened Vessel Length per 4.15.6	L	4.84	m
Distance from Saddle to Vessel tangent	a	242.00	mm
Saddle Width	b	204.00	mm
Saddle Bearing Angle	theta	120.00	degrees
Wear Plate Width	b1	306.00	mm
Wear Plate Bearing Angle	theta1	132.00	degrees
Wear Plate Thickness	tr	10.0	mm
Wear Plate Allowable Stress	Sr	186662.33	kPa
Shell Allowable Stress used in Calculation		159996.34	kPa
Head Allowable Stress used in Calculation		159996.34	kPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q, Test Case, no Ext. Forces		29029.85	N

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	23738.80	159996.34	kPa
Long. Stress at Bottom of Midspan	33081.17	159996.34	kPa
Long. Stress at Top of Saddles	28304.99	159996.34	kPa
Long. Stress at Bottom of Saddles	28468.18	159996.34	kPa
Tangential Shear in Shell	5886.89	127997.08	kPa
Circ. Stress at Horn of Saddle	2856.97	239994.52	kPa

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Circ. Compressive Stress in Shell	502.10	159996.34	kPa
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Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$\begin{aligned}
 &= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E \\
 &= 3.00 * (657.6/2 + 0) * 572.0000/647.8818 \\
 &= 870.9 \text{ N}
 \end{aligned}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$\begin{aligned}
 &= \max(F_l, \text{Friction Load}, \text{Sum of X Forces}) * B / L_s \\
 &= \max(263.23, 0.00, 0) * 572.0000/3499.9995 \\
 &= 14.2 \text{ N}
 \end{aligned}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$\begin{aligned}
 &= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st}) \\
 &= 28158 + \max(14, 870, 0, 0) \\
 &= 29029.8 \text{ N}
 \end{aligned}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	29228.05	N
Transverse Shear Load Saddle	328.80	N
Longitudinal Shear Load Saddle	86.86	N

Hydrostatic Test Pressure at center of Vessel: 2.110 N/mm²**Formulas and Substitutions for Horizontal Vessel Analysis:**

Note: Wear Plate is Welded to the Shell, k = 0.1

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0249	K8 = 0.3405
K9 = 0.2711	K10 = 0.0581	K1* = 0.1923	K6p = 0.0434
K7P = 0.0204			

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q \cdot a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a \cdot L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right] \\
 &= -29029 \cdot 0.24 \left[1 - \left(1 - \frac{0.24}{4.84} + \frac{(0.374^2 - 0.000^2)}{(2 \cdot 0.24 \cdot 4.84)} \right) / \left(1 + \frac{(4 \cdot 0.00)}{(3 \cdot 4.84)} \right) \right] \\
 &= 68.4 \text{ N-m}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= Q \cdot L / 4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) - 4a/L \\
 &= 29029 \cdot 4 / 4 \left(1 + 2 \frac{(0^2 - 0^2)}{(4^2)} \right) / \left(1 + \frac{(4 \cdot 0)}{3 \cdot 4} \right) - 4 \cdot 0 / 4 \\
 &= 28532.1 \text{ N-m}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M2 / (\pi \cdot R_m^2 t) \\
 &= 2.110 \cdot 374.055 / (2 \cdot 13.891) - 28532.1 / (\pi \cdot 374.1^2 \cdot 13.891) \\
 &= 23738.80 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M2 / (\pi \cdot R_m^2 \cdot t) \\
 &= 2.110 \cdot 374.055 / (2 \cdot 13.891) + 28532.1 / (\pi \cdot 374.1^2 \cdot 13.891) \\
 &= 33081.17 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) - M1 / (K1 \cdot \pi \cdot R_m^2 t) \\
 &= 2.110 \cdot 374.055 / (2 \cdot 13.891) - 68.4 / (0.1066 \cdot \pi \cdot 374.1^2 \cdot 13.891) \\
 &= 28304.99 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
 &= P \cdot R_m / (2t) + M1 / (K1 \cdot \pi \cdot R_m^2 \cdot t) \\
 &= 2.110 \cdot 374.055 / (2 \cdot 13.891) + 68.4 / (0.1923 \cdot \pi \cdot 374.1^2 \cdot 13.891) \\
 &= 28468.18 \text{ kPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L-2a)/(L+(4*h^2/3)) \\
 &= 29029 (4.84 - 2 * 0.24)/(4.84 + (4 * 0.00/3)) \\
 &= 26126.9 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
 &= K2 * T / (Rm * t) \\
 &= 1.1707 * 26126.86/(374.0547 * 13.8906) \\
 &= 5886.89 \text{ kPa}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 * \text{sqrt}(Rm * t) \\
 &= 0.78 * \text{sqrt}(374.055 * 13.891) \\
 &= 56.224 \text{ mm}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 * Q * k / (t * (b + X1 + X2)) \\
 &= -0.7603 * 29029 * 0.1/(13.891 * (204.00 + 56.22 + 56.22)) \\
 &= -502.10 \text{ kPa}
 \end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
 &= \min(b + 1.56 * \text{sqrt}(Rm * t), 2a) \\
 &= \min(204.00 + 1.56 * \text{sqrt}(374.055 * 13.891), 2 * 0.242) \\
 &= 316.45 \text{ mm}
 \end{aligned}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$\begin{aligned}
 &= \min(Sr/S, 1) \\
 &= \min(186662.328/159996.344, 1) \\
 &= 1.0000
 \end{aligned}$$

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$\begin{aligned}
 &= -K5 * Q * k / (B1(t + eta * tr)) \\
 &= -0.7603 * 29029 * 0.1/(316.448 (13.891 + 1.000 * 10.000)) \\
 &= -291.94 \text{ kPa}
 \end{aligned}$$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8R_m$ (4.15.27) $[\sigma_{7,r}]$:

$$= -Q / (4(t + \eta \cdot t_r) b_1) - 3K_7 Q / (2(t + \eta \cdot t_r)^2)$$

$$= -29029 / (4(13.891 + 1.000 \cdot 10.000) 316.448) -$$

$$3 \cdot 0.025 \cdot 29029 / (2(13.891 + 1.000 \cdot 10.000)^2)$$

$$= -2856.97 \text{ kPa}$$

ASME Horizontal Vessel Analysis: Stresses for the Right Saddle

(per ASME Sec. VIII Div. 2 based on the Zick method.)

Input and Calculated Values:

Vessel Mean Radius	R_m	374.05	mm
Stiffened Vessel Length per 4.15.6	L	4.84	m
Distance from Saddle to Vessel tangent	a	242.00	mm
Saddle Width	b	204.00	mm
Saddle Bearing Angle	θ	120.00	degrees
Wear Plate Width	b_1	306.00	mm
Wear Plate Bearing Angle	θ_1	132.00	degrees
Wear Plate Thickness	t_r	10.0	mm
Wear Plate Allowable Stress	S_r	95145.48	kPa
Shell Allowable Stress used in Calculation		159996.34	kPa
Head Allowable Stress used in Calculation		159996.34	kPa
Circumferential Efficiency in Plane of Saddle		1.00	
Circumferential Efficiency at Mid-Span		1.00	
Saddle Force Q , Test Case, no Ext. Forces		29054.60	N

Horizontal Vessel Analysis Results:	Actual	Allowable	

Long. Stress at Top of Midspan	23734.82	159996.34	kPa
Long. Stress at Bottom of Midspan	33085.15	159996.34	kPa
Long. Stress at Top of Saddles	28304.90	159996.34	kPa
Long. Stress at Bottom of Saddles	28468.23	159996.34	kPa

Tangential Shear in Shell	5891.91	127997.08	kPa
Circ. Stress at Horn of Saddle	3910.84	239994.52	kPa
Circ. Compressive Stress in Shell	502.53	159996.34	kPa

Intermediate Results: Saddle Reaction Q due to Wind or Seismic

Saddle Reaction Force due to Wind Ft [Fwt]:

$$\begin{aligned}
 &= F_{tr} * (F_t / \text{Num of Saddles} + Z \text{ Force Load}) * B / E \\
 &= 3.00 * (657.6/2 + 0) * 572.0000/647.8818 \\
 &= 870.9 \text{ N}
 \end{aligned}$$

Saddle Reaction Force due to Wind Fl or Friction [Fwl]:

$$\begin{aligned}
 &= \max(F_l, \text{Friction Load}, \text{Sum of X Forces}) * B / L_s \\
 &= \max(263.23, 0.00, 0) * 572.0000/3499.9995 \\
 &= 14.2 \text{ N}
 \end{aligned}$$

Load Combination Results for Q + Wind or Seismic [Q]:

$$\begin{aligned}
 &= \text{Saddle Load} + \max(F_{wl}, F_{wt}, F_{sl}, F_{st}) \\
 &= 28183 + \max(14, 870, 0, 0) \\
 &= 29054.6 \text{ N}
 \end{aligned}$$

Summary of Loads at the base of this Saddle:

Vertical Load (including saddle weight)	29252.80	N
Transverse Shear Load Saddle	328.80	N
Longitudinal Shear Load Saddle	86.86	N

Hydrostatic Test Pressure at center of Vessel: 2.110 N/mm²

Formulas and Substitutions for Horizontal Vessel Analysis:

Note: Wear Plate is Welded to the Shell, k = 0.1

The Computed K values from Table 4.15.1:

K1 = 0.1066	K2 = 1.1707	K3 = 0.8799	K4 = 0.4011
K5 = 0.7603	K6 = 0.0529	K7 = 0.0249	K8 = 0.3405

$$\begin{aligned}
 K9 &= 0.2711 & K10 &= 0.0581 & K1^* &= 0.1923 & K6p &= 0.0434 \\
 K7P &= 0.0204
 \end{aligned}$$

The suffix 'p' denotes the values for a wear plate if it exists.

Note: Dimension a is greater than or equal to $R_m / 2$.

Moment per Equation 4.15.3 [M1]:

$$\begin{aligned}
 &= -Q^*a \left[1 - \left(1 - \frac{a}{L} + \frac{(R^2 - h^2)}{(2a^*L)} \right) / \left(1 + \frac{(4h^2)}{3L} \right) \right] \\
 &= -29054^*0.24 \left[1 - \left(1 - \frac{0.24}{4.84} + \frac{(0.374^2 - 0.000^2)}{(2^*0.24^*4.84)} \right) / \left(1 + \frac{(4^*0.00)}{(3^*4.84)} \right) \right] \\
 &= 68.4 \text{ N-m}
 \end{aligned}$$

Moment per Equation 4.15.4 [M2]:

$$\begin{aligned}
 &= Q^*L/4 \left(1 + 2 \frac{(R^2 - h^2)}{(L^2)} \right) / \left(1 + \frac{(4h^2)}{(3L)} \right) - 4a/L \\
 &= 29054^*4/4 \left(1 + 2 \frac{(0^2 - 0^2)}{(4^2)} \right) / \left(1 + \frac{(4^*0)}{(3^*4)} \right) - 4^*0/4 \\
 &= 28556.4 \text{ N-m}
 \end{aligned}$$

Longitudinal Stress at Top of Shell (4.15.6) [Sigma1]:

$$\begin{aligned}
 &= P^*R_m/(2t) - M2/(\pi^*R_m^2t) \\
 &= 2.110^*374.055/(2^*13.891) - 28556.4/(\pi^*374.1^2^*13.891) \\
 &= 23734.82 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell (4.15.7) [Sigma2]:

$$\begin{aligned}
 &= P^*R_m/(2t) + M2/(\pi^*R_m^2^*t) \\
 &= 2.110^*374.055/(2^*13.891) + 28556.4/(\pi^*374.1^2^*13.891) \\
 &= 33085.15 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Top of Shell at Support (4.15.10) [Sigma*3]:

$$\begin{aligned}
 &= P^*R_m/(2t) - M1/(K1^*\pi^*R_m^2t) \\
 &= 2.110^*374.055/(2^*13.891) - 68.4/(0.1066^*\pi^*374.1^2^*13.891) \\
 &= 28304.90 \text{ kPa}
 \end{aligned}$$

Longitudinal Stress at Bottom of Shell at Support (4.15.11) [Sigma*4]:

$$\begin{aligned}
 &= P * R_m / (2t) + M1 / (K1 * \pi * R_m^2 * t) \\
 &= 2.110 * 374.055 / (2 * 13.891) + 68.4 / (0.1923 * \pi * 374.1^2 * 13.891) \\
 &= 28468.23 \text{ kPa}
 \end{aligned}$$

Maximum Shear Force in the Saddle (4.15.5) [T]:

$$\begin{aligned}
 &= Q(L - 2a) / (L + (4 * h^2 / 3)) \\
 &= 29054 (4.84 - 2 * 0.24) / (4.84 + (4 * 0.00 / 3)) \\
 &= 26149.1 \text{ N}
 \end{aligned}$$

Shear Stress in the shell no rings, not stiffened (4.15.14) [tau2]:

$$\begin{aligned}
 &= K2 * T / (R_m * t) \\
 &= 1.1707 * 26149.14 / (374.0547 * 13.8906) \\
 &= 5891.91 \text{ kPa}
 \end{aligned}$$

Decay Length (4.15.22) [x1,x2]:

$$\begin{aligned}
 &= 0.78 * \sqrt{ R_m * t } \\
 &= 0.78 * \sqrt{ 374.055 * 13.891 } \\
 &= 56.224 \text{ mm}
 \end{aligned}$$

Circumferential Stress in shell, no rings (4.15.23) [sigma6]:

$$\begin{aligned}
 &= -K5 * Q * k / (t * (b + X1 + X2)) \\
 &= -0.7603 * 29054 * 0.1 / (13.891 * (204.00 + 56.22 + 56.22)) \\
 &= -502.53 \text{ kPa}
 \end{aligned}$$

Effective reinforcing plate width (4.15.1) [B1]:

$$\begin{aligned}
 &= \min(b + 1.56 * \sqrt{ R_m * t }, 2a) \\
 &= \min(204.00 + 1.56 * \sqrt{ 374.055 * 13.891 }, 2 * 0.242) \\
 &= 316.45 \text{ mm}
 \end{aligned}$$

Wear Plate/Shell Stress ratio (4.15.29) [eta]:

$$\begin{aligned}
 &= \min(S_r / S, 1) \\
 &= \min(95145.477 / 159996.344, 1) \\
 &= 0.5947
 \end{aligned}$$

Circumferential Stress at wear plate (4.15.26) [sigma6,r]:

$$= -K5 * Q * k / (B1(t + \eta * t_r))$$

FileName : Gatti 750 heat exchanger

Horizontal Vessel Analysis (Test) : Step: 15 10:06a Dec 18, 2019

$$= -0.7603 * 29054 * 0.1 / (316.448 (13.891 + 0.595 * 10.000))$$

$$= -351.89 \text{ kPa}$$

Circ. Comp. Stress at Horn of Saddle, $L \geq 8R_m$ (4.15.27) $[\sigma_{7,r}]$:

$$= -Q / (4(t + \eta * tr) b_1) - 3 * K_7 * Q / (2(t + \eta * tr)^2)$$

$$= -29054 / (4(13.891 + 0.595 * 10.000) 316.448) -$$

$$3 * 0.025 * 29054 / (2(13.891 + 0.595 * 10.000)^2)$$

$$= -3910.84 \text{ kPa}$$

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INPUT VALUES, Nozzle Description: Noz N1 Fr10 From : 10

Pressure for Reinforcement Calculations	P	1.0000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	f	191484.98	kPa
Shell Allowable Stress At Ambient	fa	191484.98	kPa
Outside Diameter of Flat Head	Di	734.00	mm
Large Diameter of Flat Head	Dl	0.0000	mm
Flat Head Attachment Factor	F	0.41	
Head Finished (Minimum) Thickness	ea	30.0000	mm
Head Internal Corrosion Allowance	c	3.1750	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	200.0000	mm

Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa
Diameter Basis (for tr calc only)		ID	
Layout Angle		270.00	deg
Diameter		100.0000	mm.
Size and Thickness Basis		Nominal	
Nominal Thickness	eb	120	
Flange Material		SA-105	
Flange Type		Weld Neck Flange	

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Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	

Outside Projection	ho	150.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel Wgnv		11.1252	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.

Class of attached Flange	150
Grade of attached Flange	GR 1.1

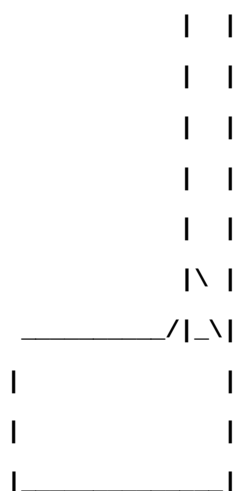
The Pressure Design option was Design Pressure + static head.

Nozzle analysis - Nozzle on the Flat Head:

Nozzle: Noz N1 Fr10 is on a Flat Head:

The analysis is to be found in the Nozzle Summary

Nozzle Sketch (may not represent actual weld type/configuration)



Abutting/Set-on Nozzle No Pad

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N1 Fr10

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N1 Fr10 Nozl: 8 10:06a Dec 18, 2019

Actual Inside Diameter Used in Calculation 92.050 mm.

Actual Thickness Used in Calculation 11.125 mm

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 114.3000. Min. thickness for this size: 8.5750 mm

Note: The minimum neck thickness chosen was to the closest table value.

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INPUT VALUES, Nozzle Description: Noz N2 Fr10 From : 10

Pressure for Reinforcement Calculations	P	1.0000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Shell Material		P355NH	
Shell Allowable Stress at Temperature	f	191484.98	kPa
Shell Allowable Stress At Ambient	fa	191484.98	kPa
Outside Diameter of Flat Head	Di	734.00	mm
Large Diameter of Flat Head	Dl	0.0000	mm
Flat Head Attachment Factor	F	0.41	
Head Finished (Minimum) Thickness	ea	30.0000	mm
Head Internal Corrosion Allowance	c	3.1750	mm
Head External Corrosion Allowance	co	0.0000	mm
Distance from Head Centerline	L1	200.0000	mm

Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa
Diameter Basis (for tr calc only)		ID	
Layout Angle		90.00	deg
Diameter		100.0000	mm.
Size and Thickness Basis		Nominal	
Nominal Thickness	eb	120	
Flange Material		SA-105	
Flange Type		Slip on	

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Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	
Outside Projection	ho	150.0000	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel Wgnv		0.0000	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.
Class of attached Flange		150	
Grade of attached Flange		GR 1.1	

The Pressure Design option was Design Pressure + static head.

Nozzle analysis - Nozzle on the Flat Head:

Nozzle: Noz N2 Fr10 is on a Flat Head:

The analysis is to be found in the Nozzle Summary

Nozzle Sketch (may not represent actual weld type/configuration)



Abutting/Set-on Nozzle No Pad

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N2 Fr10

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N2 Fr10 Nozl: 9 10:06a Dec 18, 2019

Actual Inside Diameter Used in Calculation 92.050 mm.

Actual Thickness Used in Calculation 11.125 mm

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 114.3000. Min. thickness for this size: 8.5750 mm

Note: The minimum neck thickness chosen was to the closest table value.

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INPUT VALUES, Nozzle Description: Noz N1 Fr20 From : 20

Pressure for Reinforcement Calculations	P	1.0000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Design External Pressure	Pext	0.10	N/mm ²
Temperature for External Pressure	Tempex	35	°C

Shell Material		A-106 B	
Shell Allowable Stress at Temperature	f	159996.34	kPa
Shell Allowable Stress At Ambient	fa	159996.34	kPa

Inside Diameter of Cylindrical Shell	Di	734.21	mm
Shell Finished (Minimum) Thickness	ea	13.8906	mm
Shell Internal Corrosion Allowance	c	3.1750	mm
Shell External Corrosion Allowance	co	0.0000	mm

Distance from Bottom/Left Tangent		0.1700	m
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Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa

Diameter Basis (for tr calc only)		OD	
Layout Angle		270.00	deg
Diameter		76.2000	mm.

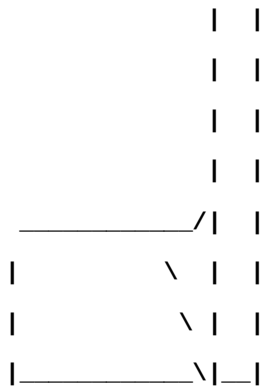
Size and Thickness Basis		Actual	
Actual Thickness	tn	7.9375	mm
Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	

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Outside Projection	ho	77.7250	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel	Wgnv	8.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle No Pad, no Inside projection

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N1 Fr20

Actual Outside Diameter Used in Calculation	76.200	mm.
Actual Thickness Used in Calculation	7.938	mm

Nozzle input data check completed without errors.

Shell/Head required thickness due to Internal Pressure (eps)

$$\begin{aligned}
 &= P * D / (2 * fs - P) + Cas + Caext \text{ [Cylindrical Shell]} \\
 &= 1.0000 * 740.5580 / (2 * 0.2E+06 - 1.0000) + 3.1750 + 0.0000 \\
 &= 5.4964 \text{ mm}
 \end{aligned}$$

Nozzle required thickness due to Internal Pressure (epb)

$$= (P * Di) / (2 * f - p) \text{ per 3.5.1.2 Eqn. 3.5.1-1}$$

$$= (1.00 * 66.6750) / (2 * 160996.33 - 1.00)$$

$$= 0.2077 + 3.1750 = 3.3827 \text{ mm}$$

Dist from Center of Nozzle to Nearest Circ. Seam: 110.000 mm

Mean noz dia / Mean Shl Dia d/D: 71.438 / 751.274 = 0.0951

Compute the value of Rho (use eas in place of ers) [rho]:

$$= d/D * \sqrt{ D / (2 * (eas - Cas + Tp)) }$$

$$= 71.44/751.27 * (751.27 / (2 * (13.89 - 3.17 + 0.00)))^{1/2}$$

$$= 0.5630$$

Compute the ratio of C eas/eps (use eas in place of ers) [CeRatio]:

$$= \text{Max}(C * (eas - Cas + Tp) / eps, 1)$$

$$= \text{Max}(1.00 * (13.89 - 3.17 + 0.00) / 2.32, 1)$$

$$= \text{Min}(4.6160, 3.10)$$

Note: The value of the CeRatio has been set to the maximum graph value.

Which is: 3.100

The following are the curves of rho selected for the analysis:

Values of Rho: 0.500 (Curve1), 0.563 (Computed rho), 0.600 (Curve2)

Values of erb/ers for values of rho: 0.500 , 0.000 , 0.600 , 0.000

Note: Since the graph line for rho cuts the erb/ers = 0 line below the calculated value of Cers/eps, this nozzle is adequately compensated.

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 76.2000. Min. thickness for this size: 7.0750 mm

Note: The minimum neck thickness chosen was to the closest table value.

3.5.4.3.4, PD 5500 Limits of Reinforcement : Case 1

Effective material Diameter limit

147.6375 mm

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N1 Fr20

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Effective material Thickness limit	Lb	3.8520	mm
Minimum Req'd Effective width adjacent to Nozzle	Ls	35.7188	mm
Thickness Limit for Nozzle Group Calculation	Tlc	18.4451	mm

The Drop for this Nozzle is : 1.9825 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 93.5981 mm

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INPUT VALUES, Nozzle Description: Noz N1 Fr40 From : 40

Pressure for Reinforcement Calculations	P	1.3000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Design External Pressure	Pext	0.10	N/mm ²
Temperature for External Pressure	Tempex	-40	°C

Shell Material		A-106 B	
Shell Allowable Stress at Temperature	f	159996.34	kPa
Shell Allowable Stress At Ambient	fa	159996.34	kPa

Inside Diameter of Cylindrical Shell	Di	734.21	mm
Shell Finished (Minimum) Thickness	ea	13.8906	mm
Shell Internal Corrosion Allowance	c	3.1750	mm
Shell External Corrosion Allowance	co	0.0000	mm

Distance from Bottom/Left Tangent		0.4800	m
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Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa

Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		139.7000	mm.

Size and Thickness Basis		Actual	
Actual Thickness	tn	12.7000	mm
Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N1 Fr40

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Outside Projection	ho	112.6500	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel Wgnv		8.0000	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Abutting/Set-on Nozzle No Pad

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N1 Fr40

Actual Outside Diameter Used in Calculation	139.700	mm.
Actual Thickness Used in Calculation	12.700	mm

Nozzle input data check completed without errors.

Shell/Head required thickness due to Internal Pressure (eps)

$$\begin{aligned}
 &= P * D / (2 * fs - P) + Cas + Caext \text{ [Cylindrical Shell]} \\
 &= 1.3000 * 740.5580 / (2 * 0.2E+06 - 1.3000) + 3.1750 + 0.0000 \\
 &= 6.1957 \text{ mm}
 \end{aligned}$$

Nozzle required thickness due to Internal Pressure (epb)

$$= (P * Di) / (2 * f - p) \text{ per 3.5.1.2 Eqn. 3.5.1-1}$$

$$= (1.30 * 120.6500) / (2 * 160996.33 - 1.30)$$

$$= 0.4891 + 3.1750 = 3.6641 \text{ mm}$$

Dist from Center of Nozzle to Nearest Circ. Seam: 200.000 mm

Mean noz dia / Mean Shl Dia d/D: 130.175 / 751.274 = 0.1733

Compute the value of Rho (use eas in place of ers) [rho]:

$$= d/D * \sqrt{ D / (2 * (eas - Cas + Tp)) }$$

$$= 130.18/751.27 * (751.27 / (2 * (13.89 - 3.17 + 0.00)))^{1/2}$$

$$= 1.0259$$

Compute the ratio of C eas/eps (use eas in place of ers) [CeRatio]:

$$= \text{Max}(C * (eas - Cas + Tp) / eps, 1)$$

$$= \text{Max}(1.00 * (13.89 - 3.17 + 0.00) / 3.02, 1)$$

$$= \text{Min}(3.5474, 3.10)$$

Note: The value of the CeRatio has been set to the maximum graph value.

Which is: 3.100

The following are the curves of rho selected for the analysis:

Values of Rho: 1.000 (Curve1), 1.026 (Computed rho), 1.200 (Curve2)

Values of erb/ers for values of rho: 1.000 , 0.000 , 1.200 , 0.000

Note: Since the graph line for rho cuts the erb/ers = 0 line below the calculated value of Cers/eps, this nozzle is adequately compensated.

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 139.7000. Min. thickness for this size: 8.5750 mm

Note: The minimum neck thickness chosen was to the closest table value.

3.5.4.3.4, PD 5500 Limits of Reinforcement : Case 1

Effective material Diameter limit

269.8750 mm

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N1 Fr40

Nozl: 11 10:06a Dec 18, 2019

Effective material Thickness limit	Lb	7.9789	mm
Minimum Req'd Effective width adjacent to Nozzle	Ls	65.0875	mm
Thickness Limit for Nozzle Group Calculation	Tlc	35.2125	mm

The Drop for this Nozzle is : 6.7065 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 133.2472 mm

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INPUT VALUES, Nozzle Description: Noz N2 Fr40 From : 40

Pressure for Reinforcement Calculations	P	1.3000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Design External Pressure	Pext	0.10	N/mm ²
Temperature for External Pressure	Tempex	-40	°C

Shell Material		A-106 B	
Shell Allowable Stress at Temperature	f	159996.34	kPa
Shell Allowable Stress At Ambient	fa	159996.34	kPa

Inside Diameter of Cylindrical Shell	Di	734.21	mm
Shell Finished (Minimum) Thickness	ea	13.8906	mm
Shell Internal Corrosion Allowance	c	3.1750	mm
Shell External Corrosion Allowance	co	0.0000	mm

Distance from Bottom/Left Tangent		0.7300	m
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Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa

Diameter Basis (for tr calc only)		OD	
Layout Angle		90.00	deg
Diameter		38.1000	mm.

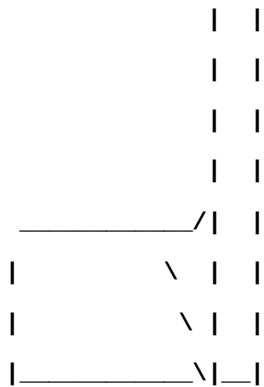
Size and Thickness Basis		Actual	
Actual Thickness	tn	8.3820	mm
Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	

Nozl: 12 10:06a Dec 18, 2019

Outside Projection	ho	39.6250	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel Wgnv		8.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle No Pad, no Inside projection

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N2 Fr40

Actual Outside Diameter Used in Calculation	38.100	mm.
Actual Thickness Used in Calculation	8.382	mm

Nozzle input data check completed without errors.

Shell/Head required thickness due to Internal Pressure (eps)

$$\begin{aligned}
 &= P * D / (2 * fs - P) + Cas + Caext \text{ [Cylindrical Shell]} \\
 &= 1.3000 * 740.5580 / (2 * 0.2E+06 - 1.3000) + 3.1750 + 0.0000 \\
 &= 6.1957 \text{ mm}
 \end{aligned}$$

Nozzle required thickness due to Internal Pressure (epb)

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N2 Fr40

Nozl: 12 10:06a Dec 18, 2019

$$= (P * Di) / (2 * f - p) \text{ per 3.5.1.2 Eqn. 3.5.1-1}$$

$$= (1.30 * 27.6860) / (2 * 160996.33 - 1.30)$$

$$= 0.1122 + 3.1750 = 3.2872 \text{ mm}$$

Dist from Center of Nozzle to Nearest Circ. Seam: 450.000 mm

Mean noz dia / Mean Shl Dia d/D: 32.893 / 751.274 = 0.0438

Compute the value of Rho (use eas in place of ers) [rho]:

$$= d/D * \sqrt{D / (2 * (eas - Cas + Tp))}$$

$$= 32.89/751.27 * (751.27 / (2 * (13.89 - 3.17 + 0.00)))^{1/2}$$

$$= 0.2592$$

Compute the ratio of C eas/eps (use eas in place of ers) [CeRatio]:

$$= \text{Max}(C * (eas - Cas + Tp) / eps, 1)$$

$$= \text{Max}(1.00 * (13.89 - 3.17 + 0.00) / 3.02, 1)$$

$$= \text{Min}(3.5474, 3.10)$$

Note: The value of the CeRatio has been set to the maximum graph value.

Which is: 3.100

The following are the curves of rho selected for the analysis:

Values of Rho: 0.200 (Curve1), 0.259 (Computed rho), 0.300 (Curve2)

Values of erb/ers for values of rho: 0.200 , 0.000 , 0.300 , 0.000

Note: Since the graph line for rho cuts the erb/ers = 0 line below the
calculated value of Cers/eps, this nozzle is adequately compensated.

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 38.1000. Min. thickness for this size: 6.2750 mm

Note: The minimum neck thickness chosen was to the closest table value.

3.5.4.3.4, PD 5500 Limits of Reinforcement : Case 1

Effective material Diameter limit

70.9930 mm

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N2 Fr40

Nozl: 12 10:06a Dec 18, 2019

Effective material Thickness limit	Lb	1.9213	mm
Minimum Req'd Effective width adjacent to Nozzle	Ls	16.4465	mm
Thickness Limit for Nozzle Group Calculation	Tlc	13.0872	mm

The Drop for this Nozzle is : 0.4946 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 54.0102 mm

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INPUT VALUES, Nozzle Description: Noz N3 Fr40 From : 40

Pressure for Reinforcement Calculations	P	1.3000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Design External Pressure	Pext	0.10	N/mm ²
Temperature for External Pressure	Tempex	-40	°C

Shell Material		A-106 B	
Shell Allowable Stress at Temperature	f	159996.34	kPa
Shell Allowable Stress At Ambient	fa	159996.34	kPa

Inside Diameter of Cylindrical Shell	Di	734.21	mm
Shell Finished (Minimum) Thickness	ea	13.8906	mm
Shell Internal Corrosion Allowance	c	3.1750	mm
Shell External Corrosion Allowance	co	0.0000	mm

Distance from Bottom/Left Tangent		2.7800	m
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Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa

Diameter Basis (for tr calc only)		OD	
Layout Angle		270.00	deg
Diameter		76.2000	mm.

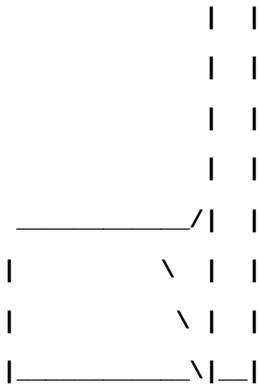
Size and Thickness Basis		Actual	
Actual Thickness	tn	7.9375	mm
Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	

Nozl: 13 10:06a Dec 18,2019

Outside Projection	ho	77.7250	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel Wgnv		8.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle No Pad, no Inside projection

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N3 Fr40

Actual Outside Diameter Used in Calculation	76.200	mm.
Actual Thickness Used in Calculation	7.938	mm

Nozzle input data check completed without errors.

Shell/Head required thickness due to Internal Pressure (eps)

$$= P * D / (2 * fs - P) + Cas + Caext \text{ [Cylindrical Shell]}$$
$$= 1.3000 * 740.5580 / (2 * 0.2E+06 - 1.3000) + 3.1750 + 0.0000$$
$$= 6.1957 \text{ mm}$$

Nozzle required thickness due to Internal Pressure (epb)

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N3 Fr40

Nozl: 13 10:06a Dec 18, 2019

$$= (P * D_i) / (2 * f - p) \text{ per 3.5.1.2 Eqn. 3.5.1-1}$$

$$= (1.30 * 66.6750) / (2 * 160996.33 - 1.30)$$

$$= 0.2703 + 3.1750 = 3.4453 \text{ mm}$$

Dist from Center of Nozzle to Nearest Circ. Seam: 2340.000 mm

Mean noz dia / Mean Shl Dia d/D: 71.438 / 751.274 = 0.0951

Compute the value of Rho (use eas in place of ers) [rho]:

$$= d/D * \sqrt{D / (2 * (eas - Cas + T_p))}$$

$$= 71.44/751.27 * (751.27 / (2 * (13.89 - 3.17 + 0.00)))^{1/2}$$

$$= 0.5630$$

Compute the ratio of C eas/eps (use eas in place of ers) [CeRatio]:

$$= \text{Max}(C * (eas - Cas + T_p) / \text{eps}, 1)$$

$$= \text{Max}(1.00 * (13.89 - 3.17 + 0.00) / 3.02 , 1)$$

$$= \text{Min}(3.5474 , 3.10)$$

Note: The value of the CeRatio has been set to the maximum graph value.

Which is: 3.100

The following are the curves of rho selected for the analysis:

Values of Rho: 0.500 (Curve1), 0.563 (Computed rho), 0.600 (Curve2)

Values of erb/ers for values of rho: 0.500 , 0.000 , 0.600 , 0.000

Note: Since the graph line for rho cuts the erb/ers = 0 line below the calculated value of Cers/eps, this nozzle is adequately compensated.

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 76.2000. Min. thickness for this size: 7.0750 mm

Note: The minimum neck thickness chosen was to the closest table value.

3.5.4.3.4, PD 5500 Limits of Reinforcement : Case 1

Effective material Diameter limit

147.6375 mm

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N3 Fr40

Nozl: 13 10:06a Dec 18, 2019

Effective material Thickness limit	Lb	4.3940	mm
Minimum Req'd Effective width adjacent to Nozzle	Ls	35.7188	mm
Thickness Limit for Nozzle Group Calculation	Tlc	18.4451	mm

The Drop for this Nozzle is : 1.9825 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 93.5981 mm

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INPUT VALUES, Nozzle Description: Noz N4 Fr40 From : 40

Pressure for Reinforcement Calculations	P	1.3000	N/mm ²
Temperature for Internal Pressure	Temp	-40	°C
Design External Pressure	Pext	0.10	N/mm ²
Temperature for External Pressure	Tempex	-40	°C

Shell Material		A-106 B	
Shell Allowable Stress at Temperature	f	159996.34	kPa
Shell Allowable Stress At Ambient	fa	159996.34	kPa

Inside Diameter of Cylindrical Shell	Di	734.21	mm
Shell Finished (Minimum) Thickness	ea	13.8906	mm
Shell Internal Corrosion Allowance	c	3.1750	mm
Shell External Corrosion Allowance	co	0.0000	mm

Distance from Bottom/Left Tangent		4.8300	m
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Type of Element Connected to the Shell : Nozzle

Material		A-106 B	
Allowable Stress at Temperature	fn	160996.33	kPa
Allowable Stress At Ambient	fna	160996.33	kPa

Diameter Basis (for tr calc only)		OD	
Layout Angle		270.00	deg
Diameter		57.1500	mm.

Size and Thickness Basis		Actual	
Actual Thickness	tn	7.4930	mm
Corrosion Allowance	can	3.1750	mm
Loading Factor applied to ers/eps	C	1.00	

FileName : Gatti 750 heat exchanger

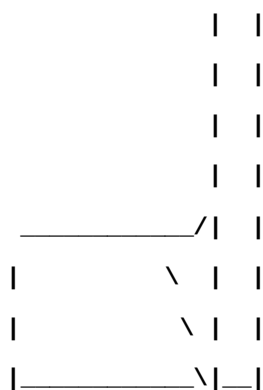
Nozzle Calcs. : Noz N4 Fr40

Nozl: 14 10:06a Dec 18, 2019

Outside Projection	ho	58.6750	mm
Weld leg size between Nozzle and Pad/Shell	Wo	10.0000	mm
Groove weld depth between Nozzle and Vessel Wgnv		8.0000	mm
Inside Projection	h	0.0000	mm
Weld leg size, Inside Element to Shell	Wi	0.0000	mm
User Defined Nozzle/Shell Centerline Angle		90.0000	deg.

The Pressure Design option was Design Pressure + static head.

Nozzle Sketch (may not represent actual weld type/configuration)



Insert/Set-in Nozzle No Pad, no Inside projection

Isolated Nozzle Calculation per PD 5500 3.5.4, Description: Noz N4 Fr40

Actual Outside Diameter Used in Calculation	57.150	mm.
Actual Thickness Used in Calculation	7.493	mm

Nozzle input data check completed without errors.

Shell/Head required thickness due to Internal Pressure (eps)

$$\begin{aligned}
 &= P * D / (2 * fs - P) + Cas + Caext \text{ [Cylindrical Shell]} \\
 &= 1.3000 * 740.5580 / (2 * 0.2E+06 - 1.3000) + 3.1750 + 0.0000 \\
 &= 6.1957 \text{ mm}
 \end{aligned}$$

Nozzle required thickness due to Internal Pressure (epb)

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N4 Fr40

Nozl: 14 10:06a Dec 18, 2019

$$= (P * D_i) / (2 * f - p) \text{ per 3.5.1.2 Eqn. 3.5.1-1}$$

$$= (1.30 * 48.5140) / (2 * 160996.33 - 1.30)$$

$$= 0.1967 + 3.1750 = 3.3717 \text{ mm}$$

Dist from Center of Nozzle to Nearest Circ. Seam: 290.000 mm

Mean noz dia / Mean Shl Dia d/D: 52.832 / 751.274 = 0.0703

Compute the value of Rho (use eas in place of ers) [rho]:

$$= d/D * \sqrt{D / (2 * (eas - Cas + T_p))}$$

$$= 52.83/751.27 * (751.27 / (2 * (13.89 - 3.17 + 0.00)))^{1/2}$$

$$= 0.4164$$

Compute the ratio of C eas/eps (use eas in place of ers) [CeRatio]:

$$= \text{Max}(C * (eas - Cas + T_p) / \text{eps}, 1)$$

$$= \text{Max}(1.00 * (13.89 - 3.17 + 0.00) / 3.02 , 1)$$

$$= \text{Min}(3.5474 , 3.10)$$

Note: The value of the CeRatio has been set to the maximum graph value.

Which is: 3.100

The following are the curves of rho selected for the analysis:

Values of Rho: 0.400 (Curve1), 0.416 (Computed rho), 0.500 (Curve2)

Values of erb/ers for values of rho: 0.400 , 0.000 , 0.500 , 0.000

Note: Since the graph line for rho cuts the erb/ers = 0 line below the
calculated value of Cers/eps, this nozzle is adequately compensated.

Minimum Nozzle Neck thickness per Table 3.5-4, mm

Branch Nominal Size: 57.1500. Min. thickness for this size: 6.7750 mm

Note: The minimum neck thickness chosen was to the closest table value.

3.5.4.3.4, PD 5500 Limits of Reinforcement : Case 1

Effective material Diameter limit

109.9820 mm

FileName : Gatti 750 heat exchanger

Nozzle Calcs. : Noz N4 Fr40

Nozl: 14 10:06a Dec 18, 2019

Effective material Thickness limit	Lb	3.2233	mm
Minimum Req'd Effective width adjacent to Nozzle	Ls	26.4160	mm
Thickness Limit for Nozzle Group Calculation	Tlc	15.1039	mm

The Drop for this Nozzle is : 1.1138 mm

The Cut Length for this Nozzle is, Drop + Ho + H + T : 73.6794 mm

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Nozzle Schedule:

Description	Nominal Flange			Noz.		Wall		Re-Pad		Cut
	Size	Sch/Type		O/Dia	Thk			ODia	Thick	Length
	mm	Cls		mm	mm			mm	mm	mm

Noz N2 Fr40	38	-	None	38.100	8.382	-	-	-	-	54
Noz N4 Fr40	57	-	None	57.150	7.493	-	-	-	-	73
Noz N1 Fr20	76	-	None	76.200	7.938	-	-	-	-	93
Noz N3 Fr40	76	-	None	76.200	7.938	-	-	-	-	93
Noz N1 Fr10	100	120	WNF	114.300	11.125	-	-	-	-	0
Noz N2 Fr10	100	120	SlipOn	114.300	11.125	-	-	-	-	0
Noz N1 Fr40	139	-	None	139.700	12.700	-	-	-	-	133

General Notes for the above table:

The Cut Length is the Outside Projection + Inside Projection + Drop +

In Plane Shell Thickness. This value does not include weld gaps,

nor does it account for shrinkage.

In the case of Oblique Nozzles, the Outside Diameter must

be increased. The Re-Pad WIDTH around the nozzle is calculated as follows:

Width of Pad = (Pad Outside Dia. (per above) - Nozzle Outside Dia.)/2

For hub nozzles, the thickness and diameter shown are those of the smaller

and thinner section.

Nozzle Material and Weld Fillet Leg Size Details:

		Shl Grve	Noz Shl/Pad	Pad OD	Pad Grve	Inside
Nozzle	Material	Weld	Weld	Weld	Weld	Weld
		mm	mm	mm	mm	mm

Noz N2	A-106 B	8.000	10.000	-	-	-
Noz N4	A-106 B	8.000	10.000	-	-	-
Noz N1	A-106 B	8.000	10.000	-	-	-

FileName : Gatti 750 heat exchanger

Nozzle Schedule : Step: 23 10:06a Dec 18, 2019

Noz N3	A-106 B	8.000	10.000	-	-	-
Noz N1	A-106 B	11.125	10.000	-	-	-
Noz N2	A-106 B	0.000	10.000	-	-	-
Noz N1	A-106 B	8.000	10.000	-	-	-

Note: The Outside projections below do not include the flange thickness.

Nozzle Miscellaneous Data:

Nozzle	Elevation/Distance	Layout	Projection		Installed In
	From Datum	Angle	Outside	Inside	Component
	m	deg.	mm	mm	

Noz N2 Fr40	0.716	90.00	39.62	0.00	Main Shell
Noz N4 Fr40	4.816	270.00	58.67	0.00	Main Shell
Noz N1 Fr20	0.120	270.00	77.72	0.00	Left Channel
Noz N3 Fr40	2.766	270.00	77.72	0.00	Main Shell
Noz N1 Fr10		270.00	150.00	0.00	Left Head
Noz N2 Fr10		90.00	150.00	0.00	Left Head
Noz N1 Fr40	0.466	90.00	112.65	0.00	Main Shell

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Nozzle Calculation Summary:

Per Section 3.5.4 Based on Graphs 3.5-9 to 3.5-11:

Description	Act Thk mm	Calc Thk mm	Table 3.5-4	Rho Result	Pr. Area Result

Noz N1 Fr10	11.13	...	8.58		
Noz N2 Fr10	11.13	...	8.58		
Noz N1 Fr20	7.94	...	7.08	Passed	N/A
Noz N1 Fr40	12.70	...	8.58	Passed	N/A
Noz N2 Fr40	8.38	...	6.28	Passed	N/A
Noz N3 Fr40	7.94	...	7.08	Passed	N/A
Noz N4 Fr40	7.49	...	6.77	Passed	N/A

Check the Spatial Relationship between the Nozzles

From Node	Nozzle Description	X Coordinate,	Layout Angle,	Mean Radius
10	Noz N1 Fr10	0.000	270.000	0.000
10	Noz N2 Fr10	0.000	90.000	0.000
20	Noz N1 Fr20	170.000	270.000	35.719
40	Noz N1 Fr40	480.000	90.000	65.088
40	Noz N2 Fr40	730.000	90.000	16.446
40	Noz N3 Fr40	2780.000	270.000	35.719
40	Noz N4 Fr40	4830.000	270.000	26.416

If any interferences/violations are found, they will be noted below.

[No interference violations have been detected !](#)

Checking Multiple Nozzles on Flat Head per PD 5500 Paragraph 3.5.5.3.1(b):Comparing Nozzles on Element: Left Head

Step: 24 10:06a Dec 18,2019

Corroded Diameter of Flat Head	734.0000	mm
Actual Corroded Head Thickness	26.8250	mm
Required Unpierced Thickness of this Flat Head	25.1102	mm

[See below for Computed Thickness of Pierced Flat Head](#)

[Note: Min. Ligament should be not less than Smaller \(of the pair\)](#)
[of Nozzles OD/4. This is not stated explicitly in the code.](#)

Sample Calculation for the First Pair: Noz N1 Fr10 & Noz N2

Nozzle	O/Dia	Wall	Corr.	ho	hi	Stress	Angle	Offset
	mm	mm	mm	mm	mm	kPa	Deg	mm

Noz N1	114.300	11.125	3.175	150.000	0.000	160996.3	270.	200.000
Noz N2	114.300	11.125	3.175	150.000	0.000	160996.3	90.	200.000

Nozzle	Inserted/Set-On

Noz N1	Set-On Type Nozzle
Noz N2	Set-On Type Nozzle

Head Factor [C1]:

$$= 0.410000$$

Distance Between the Nozzle Centers [j]:

$$\begin{aligned} x1 &= 0.0000 \quad y1 = -200.0000 \quad x2 = 0.0000 \quad y2 = 200.0000 \text{ mm} \\ j &= \text{Sqrt}((x2^2 - x1^2) + (y2^2 - y1^2)) \\ &= \text{Sqrt}((0.000^2 - -200.000^2) + (0.000^2 - 200.000^2)) \\ &= 400.0000 \text{ mm} \end{aligned}$$

Required Thickness for Each Nozzle1 [eb]:

$$\begin{aligned} 1 &= P \cdot di / (2 \cdot fn - P) = 1.000 \cdot 98.400 / (2 \cdot 160996.3 - 1.000) \\ &= 0.30653 \text{ mm} \\ 2 &= P \cdot di / (2 \cdot fn - P) = 1.000 \cdot 98.400 / (2 \cdot 160996.3 - 1.000) \\ &= 0.30653 \text{ mm} \end{aligned}$$

Credit Height for each Nozzle Outside [ho]:

$$1 = 0.8 * \text{Sqrt}((di+eb)*eb) = 0.8 * \text{Sqrt}((98.400 + 7.644) * 7.644) = 22.7763 \text{ mm}$$

$$\text{but } L = \text{Min}(L, ho) = \text{Min}(27.970, 150.000) = 27.9696 \text{ mm}$$

$$2 = 0.8 * \text{Sqrt}((di+eb)*eb) = 0.8 * \text{Sqrt}((98.400 + 7.644) * 7.644) = 22.7763 \text{ mm}$$

$$\text{but } L = \text{Min}(L, ho) = \text{Min}(27.970, 150.000) = 27.9696 \text{ mm}$$

Credit Height for each Nozzle Inside [hi]:

$$1 = 0.8 * \text{Sqrt}((di+ebi)*ebi) = 0.8 * \text{Sqrt}((98.400 + 4.775) * 4.775) = 0.0000 \text{ mm}$$

$$\text{but } Lb = \text{Min}(Lb, hi) = \text{Min}(0.000, 0.000) = 0.0000 \text{ mm}$$

$$2 = 0.8 * \text{Sqrt}((di+ebi)*ebi) = 0.8 * \text{Sqrt}((98.400 + 4.775) * 4.775) = 0.0000 \text{ mm}$$

$$\text{but } Lb = \text{Min}(Lb, hi) = \text{Min}(0.000, 0.000) = 0.0000 \text{ mm}$$

Areas Credited each Nozzle [A]:

$$1 = L*eb + Lb*(eb-c) = 27.970 * 7.644 + 0.000 * 4.775 = 0.0002 \text{ m}^2$$

$$2 = L*eb + Lb*(eb-c) = 27.970 * 7.644 + 0.000 * 4.775 = 0.0002 \text{ m}^2$$

Adjust Area for each Nozzle [A']:

$$1 = \text{Min}(A, A*fn/f) = \text{Min}(0.000, 0.000 * 160996/191484) = 0.0002 \text{ m}^2$$

$$2 = \text{Min}(A, A*fn/f) = \text{Min}(0.000, 0.000 * 160996/191484) = 0.0002 \text{ m}^2$$

Compute the Effective Diameter for Each Nozzle [d]:

$$1 = di - 2 * A'/e = 98.400 - 2 * 0.000/26.825 = 84.9979 \text{ mm}$$

$$2 = di - 2 * A'/e = 98.400 - 2 * 0.000/26.825 = 84.9979 \text{ mm}$$

Compute the Mean of Pair of Nozzles [d]:

$$= (d1 + d2)/2 = (84.998 + 84.998)/2 = 84.9979 \text{ mm}$$

Compute the Nozzle Factor [Y1]:

$$= \text{Min}(2, (j/(j-d))^{0.33})$$

$$= \text{Min}(2, (400.000/(400.000 - 84.998))^{0.33}) = 1.0829$$

Compute the Nozzle Factor [Y2]:

$$= \text{sqrt}(j/(j-d)) = \text{sqrt}(400.000/(400.000 - 84.998)) = 1.1269$$

Compute Head Thickness for Nozzle Reinforcement [e]:

FileName : Gatti 750 heat exchanger

Nozzle Summary :

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$$= \text{Max}(Y1 * eo, 0.41 * Y2 * D * \text{Sqrt}(P / f)) + c$$

$$= \text{Max}(1.083 * 21.935, 0.41 * 1.127 * 734.000 * \text{Sqrt}(1.000/191484.984)) + 3.175$$

$$= 27.6810 \text{ mm}$$

Compute Allowable Ligament (not explicit in code) [Lig]:

$$\text{Allowable} = \text{Min}(d1, d2)/4 = \text{Min}(114.300, 114.300)/4 = 28.5750 \text{ mm}$$

$$\text{Actual} = 285.7000 \text{ mm}$$

EN 13445 Paragraph 10.6.2 Ligament Checks for Nozzle Pairs :

Nozzle	Equivanlent	Y1		Y2		j	
Pair Description	Noz. Dia.						
	mm					mm	

Noz N1 Fr10 & Noz N2	84.998	1.0829		1.1269		400.00	

EN 13445 Paragraph 10.6.2 Head Required Thickness for Nozzle Pairs :

Nozzle	Ligament	Ligament	Req. thk.	Actual thk.	
Pair Description	Width	Allowable			
	mm	mm	mm	mm	

Noz N1 Fr10 & Noz N2	285.70	28.575	27.681	30.000	

Minimum Required Flat Head Thickness for Nozzle Reinforcement: 27.6810 mm

No Multiple Nozzle spacing violations have been detected !

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Input Echo, TubeSheet Item 1, Description: Gatti

Tubesheet Design Code		PD 5500	
Shell Desc.		Main Shell	
Shell Design Pressure	Ps	1.30	N/mm ²
Shell Temperature for Internal Pressure	TEMPS	-40.00	°C
Shell Material		A-106 B	
Shell Allowable Stress at Temperature	Sos	160013.92	kPa
Shell Allowable Stress at Ambient	Sas	159996.34	kPa
Shell Thickness	Ts	13.8906	mm
Shell Internal Corrosion Allowance	Cas	3.1750	mm
Inside Diameter of Shell	Ds	734.219	mm
Mean Metal Temperature for Shell	Tm	-40.00	°C
Channel Desc.		Left Channel Shell	
Channel Design Pressure	Pc	1.00	N/mm ²
Channel Temperature for Internal Pressure	TEMPC	-40.00	°C
Channel Material		A-106 B	
Channel Material UNS Number			
Channel Allowable Stress at Temperature	Soc	160013.92	kPa
Channel Allowable Stress at Ambient	Sac	159996.34	kPa
Channel Thickness	Tc	13.8906	mm
Channel Corrosion Allowance	Cac	3.1750	mm
Inside Diameter of Channel	Dc	734.219	mm
Mean Metal Temperature for Tubes	tm	-40.00	°C
Tube Design Temperature	Tubtmp	-40.00	°C
Tube Material		A-106 B	
Tube Allowable Stress at Temperature	Sot	160996.33	kPa
Tube Allowable Stress At Ambient	Sat	160996.33	kPa
Tube Yield Stress At Operating Temperature	Syt	240994.48	kPa
Tube Wall Thickness	Tt	3.3800	mm
Tube Corrosion Allowance	Catt	0.0000	mm
Number of Tubes Holes	Ntubs	136	
Tube Layout Pattern		Triangular	

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Tube Outside Diameter	do	33.4000	mm
Tube Pitch (Center to Center Spacing)	PTube	48.0000	mm
Fillet Weld Leg	af	2.0000	mm
Tube-Tubesheet Joint Classification		b	
Tube Joint Reliability Factor	fr	0.5500	
Straight Tube Length, bet. inner tubsht faces	RL	4840.000	mm
Unsupported Tube Length for max. (k*SL)	SL	1600.0000	mm
Tube end condition corres. to span (SL)	k	1.0000	
Length of Expanded Portion of Tube	l	10.0000	mm
Tube Bundle Effective outer diameter	Do	689.000	mm
Tube hole diameter	dh	32.0000	mm
Number of Grooves in tube hole	Ngroove	2	
Area of untubed Lanes,	S	0.00	m ²

Tubesheet type: Fixed Tubesheet Exchanger

Tubesheet Design Metal Temperature	TEMP TS	-40.00	°C
Tubesheet Material		A-516 70	
Tubesheet Allowable Stress at Temperature	Sots	173329.36	kPa
Tubesheet Allowable Stress at Ambient	Sats	173329.36	kPa
Thickness of Tubesheet	Tts	36.0000	mm
Tubesheet Corr. Allowance (Shell side)	Cats	3.0000	mm
Tubesheet Corr. Allowance (Channel side)	Catc	3.0000	mm
Depth of Groove in Tube Sheet	hg	6.0000	mm
Tubesheets Clamped:		Stationary Simply / Floating Clamped	

Additional Data for Fixed Tubesheet Exchangers

Mean Metal Temperature for Tubesheet	Tshm	21.11	°C
Run Multiple Load Cases for Fixed Tubesheets		No	
Is this a Kettle-type configuration		No	

Additional Data for Tubesheets Extended as Flanges:

Outside Diameter of Flanged Portion	A	850.000	mm
-------------------------------------	---	---------	----

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Diameter of Bolt Circle	C	810.000	mm
Thickness of Extended Portion of Tubesheet	Tf	56.0000	mm
Nominal Bolt Diameter	dB	16.0000	mm
Type of Thread Series	TEMA Metric Thread		
Number of Bolts	n	24	
Bolt Material			
Bolt Allowable Stress At Temperature	Sb	226994.81	kPa
Bolt Allowable Stress At Ambient	Sa	226994.81	kPa
Weld between Flange and Shell/Channel	WLDH	3.0000	mm
Is Bolt Load Transferred to the Tubesheet		Yes	

Additional Data for Gasketed Tubesheets:

Flange Face Outside Diameter	Fod	790.000	mm
Flange Face Inside Diameter	Fid	762.000	mm
Flange Facing Sketch	Code Sketch 1a		
Gasket Outside Diameter	Go	790.000	mm
Gasket Inside Diameter	Gi	762.000	mm
Gasket Factor,	m	3.7000	
Gasket Design Seating Stress	y	22993.50	kPa
Column for Gasket Seating	Code Column II		
Gasket Thickness	tg	3.1750	mm
Full face Gasket Flange Option	Program Selects		
Tubesheet Gasket on which Side	Side	CHANNEL	

Intermediate Calculations For Tubesheets Extended As Flanges:

British Standard PD 5500:2018

Gasket Contact Width,	$N = (Goc - Gic) / 2$	14.000	mm
Basic Gasket Width,	$b0 = N / 2.0$	7.000	mm
Effective Gasket Width,	$b = \text{SQRT}(b0) * 2.5$	6.665	mm
Gasket Reaction Diameter,	$G = Go - 2.0 * b$	776.671	mm

Basic Flange and Bolt loads:

Hydrostatic End Load due to Pressure[H]:

$$= 0.785 * G * G * Peq$$

$$= 0.7854 * 776.6707 * 776.6707 * 1.0000$$

$$= 473725.562 \text{ N}$$

Contact Load on Gasket Surfaces[Hp]:

$$= 2 * b * PI * G * m * P$$

$$= 2 * 6.6647 * 3.1416 * 776.6707 * 3.7000 * 1.00$$

$$= 120325.875 \text{ N}$$

Hydrostatic End Load at Flange ID[Hd]:

$$= 0.785 * Bcor * Bcor * P$$

$$= 0.785 * 768.3500 * 768.3500 * 1.0000$$

$$= 463629.562 \text{ N}$$

Pressure Force on Flange Face[Ht]:

$$= H - Hd$$

$$= 473725 - 463629$$

$$= 10095.987 \text{ N}$$

Operating Bolt Load[Wm1]:

$$= H + Hp + H'p \text{ (cannot be } < 0 \text{)}$$

$$= (473725 + 120325 + 0)$$

$$= 594051.438 \text{ N}$$

Gasket Seating Bolt Load[Wm2]:

$$= y * b * PI * G + yPart * bPart * lp$$

$$= 22993.50 * 6.6647 * 3.141 * 776.671 + 0.00 * 0.0000 * 0.00$$

$$= 373901.906 \text{ N}$$

Required Bolt Area[Am]:

$$= \text{Maximum of } Wm1/Sb, Wm2/Sa$$

$$= \text{Maximum of } 594051 / 226994, 373901 / 226994$$

$$= 0.261645E-02 \text{ m}^2$$

Bolt Spacing Correction Factor per TEMA, PD:5500 or EN-13445 [cF]:

$$= \max(\sqrt{ \text{Deltab} / (2 * dB + 6 * e / (m + 0.5)) }; 1.0)$$

$$= \max(\sqrt{ 105.726 / (2 * 16.000 + 6 * 56.000 / (3.700 + 0.5)) }; 1.0)$$

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$$= \max(0.9716 ; 1.0)$$

$$= 1.0000$$

where the actual circumferential spacing is [Deltab]:

$$= C * \sin(\text{Pi} / n)$$

$$= 810.000 * \sin(3.142 / 24)$$

$$= 105.7262 \text{ mm}$$

Bolting Information for TEMA Metric Thread Series (Non Mandatory):

Distance Across Corners for Nuts	31.180	mm
Circular Wrench End Diameter	a	0.000 mm

	Minimum	Actual	Maximum
Bolt Area, m²	0.003	0.003	
Radial distance bet. hub and bolts	20.640	21.000	
Radial distance bet. bolts and the edge	20.640	20.000	
Circumferential spacing between bolts	44.450	105.726	112.000

Flange Design Bolt Load, Gasket Seating[W]:

$$= S_a * (A_m + A_b) / 2.0$$

$$= 226994.81 * (0.0026 + 0.0033) / 2.0$$

$$= 673801.50 \text{ N}$$

Gasket Seating Force[Hg]:

$$= W_{m1} - H$$

$$= 594051 - 473725$$

$$= 120325.91 \text{ N}$$

Moment Arm Calculations:

Distance to Hub Large End[R]:

$$= (C - B_{cor}) / 2 - g1_{cor}$$

$$= (810.000 - 768.350) / 2 - 0.000$$

$$= 20.8250 \text{ mm}$$

Distance to Gasket Load Reaction[hg]:

$$= (C - G) / 2.0$$

$$= (810.0000 - 776.6707) / 2.0$$

$$= 16.6647 \text{ mm}$$

Distance to Face Pressure Reaction[ht]:

$$= (R + g1cor + hg) / 2.0$$

$$= (20.8250 + 0.0000 + 16.6647) / 2.0$$

$$= 18.7448 \text{ mm}$$

Distance to End Pressure Reaction[hd]:

$$= R + (g1cor / 2.0)$$

$$= 20.8250 + (0.0000 / 2.0)$$

$$= 20.8250 \text{ mm}$$

Summary of Moments for Internal Pressure:

Loading		Force	Distance	Bolt Corr	Moment
End Pressure,	Md	463630.	20.8250	1.0000	9659. N-m
Face Pressure,	Mt	10096.	18.7448	1.0000	189. N-m
Gasket Load,	Mg	120326.	16.6647	1.0000	2006. N-m
Gasket Seating,	Ma	673802.	16.6647	1.0000	11233. N-m
Total Moment for Operation,			Mo		11854. N-m
Total Moment for Gasket Seating,			Ma		11233. N-m

Tube Properties - Group # 1.10 and S factor: 1.40

Tube alpha at Metal Temp. along len.	-40.0	0.1081E-04 /°C
Tube Elastic Mod. at Metal Temp. length	-40.0	0.2100E+06 N/mm²

Shell Properties - Group # 1.40 and S factor: 1.40

Shell alpha at Metal Temp. along len.	-40.0	0.1081E-04 /°C
Shell Elastic Mod. at Metal Temp. len	-40.0	0.2100E+06 N/mm²

Tubesheet Properties - Group # 1.40 and S factor: 1.40

Tubesheet Elastic Mod. at Design Temp. -40.0 0.2100E+06 N/mm²

Channel Properties - Group # 1.40 and S factor: 1.40

Channel Elastic Mod. at Design Temp. -40.0 0.2100E+06 N/mm²

PD 5500:2012, Paragraph 3.9 Flat Heat Exchanger Tubesheets:

Tubesheet Min Thk, for tube od between 30.000-40.000 is : 25.000 mm

$$TMIN = 25.0000 \text{ mm}$$

Min. Thickness + CATS + MAX(CATC, GROOVE)

$$TREQMIN = 34.0000 \text{ mm}$$

Paragraph 3.9.4: Tubesheets of Fixed Tubesheet Exchangers

[Note: Iterating to compute Tubesheet Required Thickness per 3.9.4.2:](#)

For the Effective Shell Side Pressure - Bending [e]:

$$\begin{aligned} &= D1 * \sqrt{PB1 / (\Omega S * \mu * f * 4 * H)} \\ &= 740.569 * \sqrt{0.361 / (2.000 * 0.353 * 173329.359 * 4 * 6.772)} \\ &= 7.735 \text{ mm} \end{aligned}$$

For the Effective Tubeside Pressure - Bending [e]:

$$\begin{aligned} &= D2 * \sqrt{PB2 / (\Omega T * \mu * f * 4 * H)} \\ &= 776.671 * \sqrt{0.567 / (2.000 * 0.353 * 173329.359 * 4 * 6.772)} \\ &= 10.166 \text{ mm} \end{aligned}$$

For the Effective Shellside Pressure - Shear [e]:

$$\begin{aligned} &= 0.155 * Do * PS1 / (\lambda * \tau) \\ &= 0.155 * 689.000 * 0.361 / (0.304 * 86664.680) \\ &= 1.463 \text{ mm} \end{aligned}$$

For the Effective Tubeside Pressure - Shear [e]:

$$\begin{aligned} &= 0.155 * Do * PS2 / (\lambda * \tau) \\ &= 0.155 * 689.000 * 0.384 / (0.304 * 86664.680) \\ &= 1.555 \text{ mm} \end{aligned}$$

Tubesheet Extension Reqd Thickness per paragraphs 3.9.3.1 & 3.9.4.1 [e1]:

$$\begin{aligned}
 &= \max((1.909 * W * hg / (G * fa))^{1/2}, (1.909 * Wm1 * hg / (G * f))^{1/2}) \\
 &= (1.909 * W * hg / (G * fa))^{1/2} \quad (\text{governing in this case}) \\
 &= (1.909 * 673801 * 16.665 / (776.671 * 173329.359))^{1/2} \\
 &= 12.619 \text{ mm}
 \end{aligned}$$

[Note: Recomputing Effective Pressures at the given thickness.](#)

Equivalent Shell Side Bolting Pressure - Para 3.9.4.3.5 [PBS]:

$$\begin{aligned}
 &= 2 * PI * MATM / D2^3 \\
 &= 2 * 3.142 * 11233 / 776.671^3 \\
 &= 0.151 \text{ N/mm}^2
 \end{aligned}$$

Equivalent Tube Side Bolting Pressure - Para 3.9.4.3.5 [PBT]:

$$\begin{aligned}
 &= 2 * PI * MOP / D1^3 \\
 &= 2 * 3.142 * 11854 / 740.569^3 \\
 &= 0.183 \text{ N/mm}^2
 \end{aligned}$$

Ligament Efficiency for Bending - paragraph 3.9.2.1 [Mu]:

$$\begin{aligned}
 &= (P^* - d^*h) / P^* \\
 &= (48.000 - 31.074) / 48.000 \\
 &= 0.353
 \end{aligned}$$

Ligament Efficiency for Shear - paragraph 3.9.2.1 [Lambda]:

$$\begin{aligned}
 &= (P - dh) / P \\
 &= (48.000 - 33.400) / 48.000 \\
 &= 0.304
 \end{aligned}$$

Factor k: para 3.9.4.2 [k]:

$$\begin{aligned}
 &= 4 * N * Et * et * (d - et) / (L * D1^2) \\
 &= 4 * 136 * 210000.0 * 3.38 (33.40 - 3.38) / (4840.00 * 740.57^2) \\
 &= 4.3665 \text{ N /mm}^3
 \end{aligned}$$

Differential Factor Kc para 3.9.4.2 [Kc]:

$$= Ec * ec^{2.5} / ((12(1 - v^2))^{0.75} * (D2 + ec)^{0.50})$$

$$= 210000.016 * 10.716^{2,5} / ((12(1 - 0.300^2))^{0.75} * (776.671 + 10.716)^{0.50})$$

$$= 468236 \text{ N}$$

Per 3.9.4.2 - There is a gasket on the tubeside thus: $K_c = 0$

Differential Factor K_s Para 3.9.4.2 [K_s]:

$$= E_s * e_s^{2.5} / ((12(1 - v^2))^{0.75} * (D_1 + e_s)^{0.50})$$

$$= 210000.016 * 10.716^{2,5} / ((12(1 - 0.300^2))^{0.75} * (740.569 + 10.716)^{0.50})$$

$$= 479354 \text{ N}$$

$$K_{\theta} = K_c + K_s$$

$$= 0 + 479354$$

$$= 479354 \text{ N}$$

Eta from Figure 3.9-7 or 3.9-8 ($e < 2P$ or $e > 2P$) [η]:

$$= 0.39201$$

D^* per paragraph 3.9.4.2 [D^*]:

$$= \eta * E * e^3 / (12(1 - v^2))$$

$$= 0.392 * 210000.016 * 27.000^3 / (12(1 - 0.300^2))$$

$$= 148431.250 \text{ N-m}$$

Beta per paragraph 3.9.4.2 [β]:

$$= (2 * k / D^*)^{0.25}$$

$$= (2 * 4.366 / 148431.250)^{0.25}$$

$$= 0.156E-01 \text{ 1/mm}$$

x_a per paragraph 3.9.4.2 [x_a]:

$$= \beta * D_1 / 2$$

$$= 0.016 * 740.569 / 2$$

$$= 5.768$$

z per paragraph 3.9.4.2 [z]:

$$= 2 * K_{\theta} / (\beta * D^*)$$

$$= 2 * 479354.688 / (0.016 * 148431.250)$$

$$= 0.415$$

K per paragraph 3.9.4.2 [K]:

$$\begin{aligned} &= E_s * e_s * (D - e_s) / (E_t * e_t * N(d - e_t)) \\ &= 210000.016 * 10.716 * (762.000 - 10.716) / \\ &\quad (136 * 33.400 * 3.380 (33.400 - 3.380)) \\ &= 0.583 \end{aligned}$$

Fq From Figure 3.9-9 (using z & xa from above) [Fq]:

$$= 3.60950$$

fs per paragraph 3.9.4.3.1 [fs]:

$$\begin{aligned} &= 1 - N * (d / D1)^2 \\ &= 1 - 136.000 * (33.400 / 740.569)^2 \\ &= 0.723 \end{aligned}$$

ft per paragraph 3.9.4.3.2 [ft]:

$$\begin{aligned} &= 1 - N * ((d - 2 * e_t) / D1)^2 \\ &= 1 - 136 * ((33.400 - 2 * 3.380) / 740.569)^2 \\ &= 0.840 \end{aligned}$$

Ps' per paragraph 3.9.4.3.1 [Ps']:

J = 1 (No Expansion Joint)

$$\begin{aligned} X &= 0.4 * J(1.5 + K(1.5 + fs)) - ((1 - J) / 2) * ((DJ / D1)^2 - 1) \\ &= 0.4 * 0.100E+01(1.5 + 0.583 (1.5 + 0.723)) - ((1 - 0.100E+01) / 2 * \\ &\quad (740.569 / 740.569)^2 - 1) \\ &= 1.118832 \end{aligned}$$

$$\begin{aligned} Ps' &= p1 * X / (1 + J * K * Fq) \\ &= 1.300 * 1.119 / (1 + 0.100E+01 * 0.583 * 3.610) \\ &= 0.468 \text{ N/mm}^2 \end{aligned}$$

Pt' per paragraph 3.9.4.3.2 [Pt']:

$$X = p2 * (1 + 0.4 * J * K * (1.5 + ft))$$

$$= 1.000 * (1 + 0.4 * 0.100E+01 * 0.583 * (1.5 + 0.840))$$

$$= 1.546047 \text{ N/mm}^2$$

$$P_t' = X / (1 + J * K * F_q)$$

$$= 1.546 / (1 + 0.100E+01 * 0.583 * 3.610)$$

$$= 0.498 \text{ N/mm}^2$$

Differential Thermal Expansion Pressure p_e per 3.9.4.3.3:

$$X = E_s * e_s * (a_s * \Theta_{sT} - a_t * \Theta_{tT})$$

$$= 210000.016 * 10.716 * (0.000011 * -50.002 - 0.000011 * -50.002)$$

$$= 0.000 \text{ N / mm}$$

$$Y = X / ((D - 3 * e_s) * (1 + J * K * F_q))$$

$$= 0.000 / ((762.000 - 3 * 10.716) * (1 + 0.100E+01 * 0.583 * 3.610))$$

$$= 0.000 \text{ mm}$$

$$p_e = 4 * J * Y$$

$$= 4 * 0.100E+01 * 0.000$$

$$= 0.000E+00 \text{ N/mm}^2$$

Calculation of Shell Side Effective Pressure per 3.9.4.3.1:

[Note: Absolute values are given below](#)

$$PSE0 = 0.667 * (p_s' - p_e)$$

$$= 0.667 * (0.468 - 0.000E+00)$$

$$= 0.312 \text{ N/mm}^2$$

$$PSE1 = p_s'$$

$$= 0.468 \text{ N/mm}^2$$

$$PSE2 = P_{bs}$$

$$= 0.151 \text{ N/mm}^2$$

$$PSE3 = 0.667 * (p_s' - p_e - P_{bs})$$

$$= 0.667 * (0.468 - 0.000E+00 - 0.151)$$

$$= 0.212 \text{ N/mm}^2$$

$$PSE4 = 0.667 * (P_{bs} + p_e)$$

$$= 0.667 * (0.151 + 0.000E+00)$$

$$= 0.100 \text{ N/mm}^2$$

$$PSE5 = p_s' - P_{bs}$$

$$= 0.468 - 0.151$$

$$= 0.318 \text{ N/mm}^2$$

Maximum shell side effective pressure from above - bending:

$$PB1 = 0.468 \text{ N/mm}^2$$

Maximum shell side effective pressure from above - shear:

$$\begin{aligned} PSE6 &= 0.667 * (Ps' - Pe) \\ &= 0.667 * (0.468 - 0.000E+00) \\ &= 0.312 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PSE7 &= Ps' \\ &= 0.468 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PSE10 &= 0.667 * Pe \\ &= 0.667 * 0.000E+00 \\ &= 0.000 \text{ N/mm}^2 \end{aligned}$$

Maximum effective pressure for shear:

$$P1S = 0.468 \text{ N/mm}^2$$

If PB1 is controlled by term containing pe then Omega = 1.5 else 2.0

$$\Omega_{S} = 2.000$$

Calculation of tube side effective pressure per 3.9.4.3.2:

[Note: Absolute values are given below](#)

When $ps' > 0$ then:

$$\begin{aligned} PTE0 &= 0.667 * (pt' + Pbt + pe) \\ &= 0.667 * (0.498 + 0.183 + 0.000E+00) \\ &= 0.454 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PTE1 &= pt' + Pbt \\ &= 0.498 + 0.183 \\ &= 0.681 \text{ N/mm}^2 \end{aligned}$$

When $ps' < 0$ then:

$$\begin{aligned} PTE2 &= 0.667(Pt' - Ps' + Pbt + Pe) \\ &= 0.667(0.498 - 0.468 + 0.183 + 0.000E+00) \\ &= 0.142 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PTE3 &= pt' - ps' + Pbt \\ &= 0.498 - 0.468 + 0.183 \end{aligned}$$

$$= 0.213 \text{ N/mm}^2$$

Maximum tube side pressure for bending:

$$PB2 = 0.681 \text{ N/mm}^2$$

Maximum Shear: eliminate the Pbs term in above equations

$$\begin{aligned} PTE4 &= 0.667(Pt' + Pe) \\ &= 0.667(0.498 + 0.000E+00) \\ &= 0.332 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PTE5 &= pt' \\ &= 0.498 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PTE6 &= 0.667(pt' - ps' + pe)' \\ &= 0.020 \text{ N/mm}^2 \end{aligned}$$

$$\begin{aligned} PTE7 &= pt' - ps' \\ &= 0.498 - 0.468 \\ &= 0.029 \text{ N/mm}^2 \end{aligned}$$

If $ps' > 0$ then $PS2 = \text{Max}(PTE4, PTE5)$ else $ps' = \text{Max}(PTE6, PTE7)$:

$$PS2 = 0.498 \text{ N/mm}^2$$

If PB2 is controlled by term containing pe then Omega = 1.5 else 2.0

$$\Omega_{\text{OmegaT}} = 2.000$$

The value of H is taken from Figure 3.9-10 or 3.9-11 using z and xa:

$$H = 5.934$$

Shell and Tube Longitudinal Stresses per paragraph 3.9.4.4:

Shell Longitudinal Stress Calculations per 3.9.5:

P* has a maximum positive and maximum negative value:

$$\text{Maximum positive } P^* = 0.971 \text{ N/mm}^2$$

$$\text{Maximum negative } P^* = 0.000E+00 \text{ N/mm}^2$$

Calculated Positive Shell Longitudinal Stress [Fpos]:

$$\begin{aligned} &= Pstar * (D - es) / (4 * es) \\ &= 0.971 * (762.000 - 10.716) / (4 * 10.716) \\ &= 16866.252 \text{ kPa} \end{aligned}$$

Max. Allowable Postive Long. Shell Stress = 160013.922 kPa

Calculated Negative Shell Longitudinal Stress [Fneg]:

$$\begin{aligned}
 &= Pstar * (D - es) / (4 * es) \\
 &= 0.000 * (762.000 - 10.716) / (4 * 10.716) \\
 &= 0.000 \text{ kPa}
 \end{aligned}$$

Maximum Allowable Negative Long. Stress, f:

$$\begin{aligned}
 pyss &= 2 * s' * fshell * es / R \\
 &= 2 * 1.400 * 160013.922 * 10.716 / 375.642 \\
 &= 12780.827 \text{ kPa}
 \end{aligned}$$

$$\begin{aligned}
 pe &= 1.21 * Es * es^2 / R^2 \\
 &= 1.21 * 210000.016 * 10.716^2 / 5555.396^2 \\
 &= 206.772 \text{ N/mm}^2
 \end{aligned}$$

$$\begin{aligned}
 K &= pe / pyss \\
 &= 206.772 / 12780.827 \\
 &= 16.177
 \end{aligned}$$

DELTA, From figure A-2:

$$= 0.551$$

$$\begin{aligned}
 fallow &= DELTA * s' * fshell \\
 &= 0.551 * 1.400 * 160013.922 \\
 fallow &= 123459.000 \text{ kPa}
 \end{aligned}$$

Tube Longitudinal Stresses:

P* has both positive and negative values:

$$P^* \text{ positive} = 0.265 \text{ kPa}$$

$$P^* \text{ negative} = -0.208 \text{ N/mm}^2$$

Calculated Positive Tube Longitudinal Stress [PosStress]:

$$\begin{aligned}
 &= (Pt^*) * Fq * D2^2 / (4 * N * (et - ct) * (d - (et - ct))) \\
 &= (0.27) * 3.61 * 776.671^2 / (4 * 136 * 3.380 * (33.400 - \\
 &\quad (3.380 - 0.000)))
 \end{aligned}$$

$$= 10455.822 \text{ kPa}$$

Maximum Allowable Positive Stress [fpos]:

$$= 160996.328 \text{ kPa}$$

Calculated Negative Tube Longitudinal Stress [NegStress]:

$$\begin{aligned} &= (Pt^*) * Fq * D^2 / (4 * N * (et - ct) * (d - (et - ct))) \\ &= (-0.21) * 3.61 * 776.671^2 / (4 * 136 * 3.380 * (33.400 - \\ &\quad (3.380 - 0.000))) \\ &= -8195.981 \text{ kPa} \end{aligned}$$

Maximum Allowable Compressive Stress :

$$\begin{aligned} C &= \text{SQRT}(2 * \text{Pi} * Et / (s' * ftube)) \text{ per para 3.9.5} \\ &= \text{SQRT}(2 * 3.142 * 210000.016 * / (1.400 * 160996.328)) \\ &= 135.609 \end{aligned}$$

$$\begin{aligned} S &= 3.25 - 0.50 * Fq \text{ per para 3.9.5 (Bet. 1.25 - 2.0)} \\ &= 3.25 - 0.50 * 3.610 \\ &= 1.445 \end{aligned}$$

$$\begin{aligned} f_{neg} &= \text{pi}^2 * r^2 * Et / (S * Lk^2) \quad (\text{When } C \leq Lk / r) \\ &= 3.142^2 * 10.681^2 * 210000.016 / (1.445 * 1600.000^2) \\ &= 63901.898 \text{ kPa} \end{aligned}$$

Allowable Load on the Tube/Tubesheet Joint, per 3.9.6 [Fallow]:

$$\begin{aligned} &= \text{AreaTube} * f * Fr \quad (\text{Jt type b Table 3.9-2}) \\ &= 0.000 * 160996.328 * 0.550 \\ &= 28225.672 \text{ N} \end{aligned}$$

The actual loading on the tubes is Max of stress * area_tube [ActualLoad]:

$$\begin{aligned} &= \text{Max}(\text{NegStress}, \text{PosStress}) * \text{area_tube} \\ &= \text{Max}(-8195.981, 10455.822) * 0.000 \\ &= 3332.912 \text{ N} \end{aligned}$$

Summary of Tubesheet Results:

Condition	Req Thk (+CA)	Actual Thk (mm)	Actual Stress	Allow (kPa)	Result
Bending	19.166	36.000	67338.64	346658.72	Ok
Shear	10.555	36.000	6473.06	86664.68	Ok
Min. Per code	34.000	36.000	-----	-----	Ok
Bolted Exten.	12.619	56.000	8801.02	173329.36	Ok

Longitudinal Stresses:

Condition	Compressive		Tensile		Result
	Actual	Allow	Actual	Allow	
Shell	0.000	*****	16866.252	160013.922	Ok
Long Tubes	-8195.981	-63901.898	10455.822	160996.328	Ok
Tube Load			3332.912	28225.672	Ok

British Standard PD 5500:2018

Diameter Spec : 762.000 mm OD

Vessel Design Length, Tangent to Tangent 5.47 m

Specified Datum Line Distance 0.05 m

Shell Material A-106 B

Shell Side Design Temperature -40 °C

Channel Side Design Temperature -40 °C

Shell Side Design Pressure 1.300 N/mm²Channel Side Design Pressure 1.000 N/mm²Shell Side Hydrostatic Test Pressure 2.106 N/mm²Channel Side Hydrostatic Test Pressure 1.620 N/mm²

Wind Design Code ASCE-93

Earthquake Design Code UBC-94

Element Pressures and MAWP: N/mm²

Element Desc	Design Pres.	External	M.A.W.P	Corrosion
	+ Stat. head	Pressure		Allowance

Left Head	1.000	0.103	1.871	3.1750
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Left Channel Shell	1.000	0.103	4.564	3.1750
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Flange 30	1.000	0.103	1.321	3.1750
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Main Shell	1.300	0.103	4.564	3.1750
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Flange 50	1.000	0.103	1.321	3.1750
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Right Channel	1.000	0.103	4.564	3.1750
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Right head	1.000	0.103	1.876	3.1750
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Element Type	"To" Elev m	Length m	Element Thk mm	R e q d Int.	T h k Ext.	Joint Eff Long	Circ
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FileName : Gatti 750 heat exchanger

Vessel Design Summary :

Step: 25 10:06a Dec 18, 2019

Wld Flat	0.0	0.0	30.0	25.1	No Calc	1.00	1.00
Cylinder	0.2	0.2	13.9	5.5	4.7	0.80	1.00
Body Flg	0.2	0.1	55.0	46.5	46.5	1.00	1.00
Cylinder	5.1	4.8	13.9	6.3	8.0	0.80	1.00
Body Flg	5.2	0.1	55.0	46.5	46.5	1.00	1.00
Cylinder	5.4	0.2	13.9	5.5	4.7	1.00	1.00
Wld Flat	5.4	0.0	30.0	25.1	No Calc	1.00	1.00

Element thicknesses are shown as Nominal if specified, otherwise are Minimum

Saddle Parameters:

Saddle Width	204.000	mm
Saddle Bearing Angle	120.000	deg.
Centerline Dimension	572.000	mm
Wear Pad Width	306.000	mm
Wear Pad Thickness	10.000	mm
Wear Pad Bearing Angle	132.000	deg.
Distance from Saddle to Tangent	242.000	mm

Summary of Maximum Saddle Loads, Operating Case :

Maximum Vertical Saddle Load	19404.81	N
Maximum Transverse Saddle Shear Load	996.37	N
Maximum Longitudinal Saddle Shear Load	263.23	N

Summary of Maximum Saddle Loads, Hydrotest Case :

Maximum Vertical Saddle Load	29252.80	N
Maximum Transverse Saddle Shear Load	328.80	N
Maximum Longitudinal Saddle Shear Load	86.86	N

Weights:

Fabricated - Bare W/O Removable Internals	3690.1	kgm
Shop Test - Fabricated + Water (Full)	5745.7	kgm
Shipping - Fab. + Rem. Intls.+ Shipping App.	3690.1	kgm
Erected - Fab. + Rem. Intls.+ Insul. (etc)	3690.1	kgm
Empty - Fab. + Intls. + Details + Wgths.	3690.1	kgm

Operating - Empty + Operating Liquid (No CA)	3690.1	kgm
Field Test - Empty Weight + Water (Full)	5745.7	kgm